Synthetic Aperture Radar Interferometry

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Abstract

Synthetic aperture radar interferometry is an imaging technique for measuring the topography of a surface, its changes over time, and other changes in the detailed characteristics of the surface. By exploiting the phase of the coherent radar signal, interferometry has transformed radar remote sensing from a largely interpretive science to a quantitative tool, with applications in cartography, geodesy, land cover characterization, and natural hazards. This paper reviews the techniques of interferometry, systems and limitations, and applications in a rapidly growing area of science and engineering.

Keywords

SAR, Synthetic Aperture Radar, Interferometry, Geophysical Applications.

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I. INTRODUCTION

This paper describes an ingenious remote sensing technique generally referred to as Interferometric Synthetic Aperture Radar (InSAR, sometimes termed IFSAR or ISAR). InSAR is the synthesis of conventional SAR techniques and interferometry techniques that have developed over several decades in radio astronomy [1]. InSAR developments in recent years have addressed some of the limitations in conventional SAR systems and subsequently have opened entirely new application areas in earth system science studies.

SAR systems have been used extensively in the past two decades for fine resolution mapping and other remote sensing applications. [2], [3], [4]. Operating at microwave frequencies, SAR systems provide unique images representing the electrical and geometrical properties of a surface in nearly all weather conditions. Since they provide their own illumination, SARs can image in daylight or at night. SAR data are increasingly applied to geophysical problems, either by themselves or in conjunction with data from other remote sensing instruments. Examples of such applications include polar ice research, land use mapping, vegetation and biomass measurements, and soil moisture mapping [3]. At present, a number of spaceborne SAR systems from several countries and space agencies are routinely generating data for such research [5].

Fig. 1. Imaging scenario for typical SAR system. The platform carrying the SAR instrument follows a rectilinear track. The radar antenna looks to the side, imaging the terrain below.

A conventional SAR only measures the along-track and cross-track location of a target. Targets which have the same along-track and cross-track locations but different altitudes will be projected to a two-dimensional imaging plane, as illustrated in Figure 2 [4]. For many applications, this altitude-dependent distortion adversely affects the interpretation of the imagery. InSAR techniques have developed to measure the third dimension.

The first report of an InSAR system was by Graham [7]. He augmented a conventional SAR system with an additional physical antenna displaced in the cross-track plane from the conventional SAR antenna, forming an imaging interferometer. By mixing the signals from the two antennas, the Graham interferometer recorded amplitude variations that represented the beat pattern of the relative

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phase of the signals. The relative phase changes with the topography of the surface as described below, so the fringe variations tracks the topographic contours.

To overcome the inherent difficulties of inverting amplitude fringes to obtain topography, subsequent InSAR systems were developed to record the complex amplitude and phase information digitally for each antenna. In this way, the relative phase of each image point could be reconstructed directly. The first demonstrations of such systems with an airborne platform was reported by Zebker and Goldstein[8], and with a spaceborne platform using SeaSAT data by Li and Goldstein [9]. Today, over a dozen airborne interferometers exist throughout the world, and interferometry using spaceborne SAR is enjoying widespread application. This review is written in recognition of this explosion in popularity and utility of this method.

Fig. 2. The three-dimensional world is collapsed to two dimensions in conventional SAR imaging.

The paper is organized to first provide an overview of the concepts of InSAR, followed by more detailed discussions on InSAR theory, system issues, and examples of applications. Section II provides a consistent mathematical representation of InSAR principles, including issues that impact processing algorithms and phenomenology associated with such InSAR data. Section III describes the implementation approach for various types of InSAR systems with descriptions of some of the specific InSAR systems which are either operational or planned in the next few years. Section IV provides a broad overview of the applications of InSAR results, including topographic mapping, ocean current measurement, glacier motion detection, earthquake and hazard mapping, and vegetation estimation and classification. Finally, Section V provides our outlook on the development and impact of InSAR in remote sensing. Table B in the Appendix lists the symbols used in the equations in this paper and their definitions.

II. Overview of Interferometric SAR

A. Interferometry for Topography

Figure 3 illustrates the InSAR system concept. While radar pulses are transmitted from the conventional SAR antenna, radar echoes are received by both the conventional and an additional SAR antenna. By coherently combining the signals from the two antennas, the interferometric phase difference between the received signals can be formed for each imaged point. In this scenario, the phase difference is essentially related to the geometric path length difference to the image point, which depends on the topography. With knowledge of the interferometer geometry, the phase difference can be converted into an altitude for each image point. In essence, the phase difference provides a third measurement, in addition to the along and cross track location of the image point, or "target," to allow a reconstruction of the three-dimensional location of the targets.

The InSAR approach for topographic mapping is similar in principle to the conventional stereoscopic approach. In stereoscopy, a pair of images of the terrain are obtained from two displaced imaging positions. The 'parallax' obtained from the displacement allows the retrieval of topography because targets at different heights are displaced relative to each other in the two images by an amount related to the their altitudes [10].

The major difference between the InSAR technique and stereoscopy is that the 'parallax' measurements between the SAR images are obtained by measuring the phase difference between the signals received by two InSAR antennas. As illustrated in Fig. 3, these phase differences can be used to determine the angle of the target relative to the baseline of the interferometric SAR directly. The accuracy of the InSAR parallax measurement is typically a fraction of the SAR wavelength (say, several cm) whereas the parallax measurement accuracy of the stereoscopic approach is usually a fraction of the resolution of the imagery (say, several m).

Typically, the post spacing of the InSAR topographic data are comparable to the fine spatial resolution of SAR imagery while the altitude measurement accuracy generally exceeds stereoscopic accuracy at comparable resolutions. The registration of the two SAR images for the interferometric measurement, the retrieval of the interferometric phase difference and subsequent conversion of the results into digital elevation models of the terrain can be highly automated, an intrinsic advantage of the InSAR approach. As discussed in the sections below, the performance of InSAR systems is largely understood both theoretically and experimentally. These developments have led to airborne and spaceborne InSAR systems for routine topographic mapping.

Fig. 3. Interferometric SAR for topographic mapping uses two apertures separated by a "baseline" to image the surface. The phase difference between the apertures for each image point, along with the range and knowledge of the baseline, can be used to infer the precise shape of the imaging triangle to derive the topographic height of the image point.

B. Interferometry for Surface Change

Another interferometric SAR technique was advanced by Goldstein and his associates [11] for measurement of surface motion by imaging the surface at multiple times (Fig. 4). The time separation between the imaging can be a fraction of a second to years. The multiple images can be thought of as "time-lapse" imagery. A target movement will be detected by comparing the images. Unlike conventional schemes in which motion is detected only when the targets move more than a significant fraction of the resolution of the imagery, this technique measures the phase differences of the pixels in each pair of the multiple SAR images. If the flight path and imaging geometries of all the SAR observations are identical, any interferometric phase difference is due to changes over time of the SAR system clock, variable propagation delay, or surface motion in the direction of the radar line of sight.

In the first application of this technique described in the open literature, Goldstein and Zebker [11] augmented a conventional airborne SAR system with an additional aperture, separated along the length of the aircraft fuselage from the conventional SAR antenna. Given an antenna separation of roughly 20 m and an aircraft speed of about 200 m/s, the time between target observations made by the two antennas was about 10 ms. Over this time interval clock drift and propagation delay variations are negligible. Goldstein and Zebker showed that this system was capable of measuring tidal motions in the San Francisco bay area with an accuracy of several cm/s. This technique has been dubbed "along-track interferometry" (ATI) because of the arrangement of two antennas along the flight track on a single platform. In the ideal case, there is no cross-track separation of the apertures, and therefore no sensitivity to topography.

Fig. 4. An along track interferometer maintains a baseline separated along the flight track such that surface points are imaged by each aperture within one second. Motion of the surface over the elapsed time is recorded in the phase difference of the pixels.

C. General Interferometry: Topography and Change

ATI is merely a special case of "repeat-track interferometry" (RTI), which can been used to generate topography and motion. The orbits of several spaceborne SAR satellites have been controlled in such a way that they nearly retrace themselves after several days. Aircraft can also be controlled to accurately repeat flight paths. If the repeat flight paths result in a cross-track separation and the surface has not changed between observations, then the repeat-track observation pair can act as an interferometer for topography measurement.

If the flight track is repeated perfectly such that there is no cross-track separation, then there is no sensitivity to topography, and radial motions can be measured directly as with an ATI system.

Since the temporal separation between the observations is typically hours to days, however, the ability to detect small radial velocities is substantially better than the ATI system described above. The first demonstration of repeat track interferometry for velocity mapping was a study of the Rutford ice stream in Antarctica, again by Goldstein and colleagues [12]. The radar aboard the European Space Agency's ERS-1 satellite obtained several SAR images of the ice stream with near-perfect retracing so that there was no topographic signature in the interferometric phase. Goldstein *et al.* showed that measurements of the ice stream flow velocity of the order of 1 m/yr (or 3×10^{-8} m/s) can be obtained using observations separated by a few days.

Most commonly for repeat-track observations, the track of the sensor does not repeat itself exactly, so the interferometric time-separated measurements generally comprise the signature of topography and radial motion or surface displacement. The approach for reducing these data into velocity or surface displacement by removing topography is generally referred to as "differential interferometric SAR." In this approach (Fig. 5), at least three images are required to form two interferometric phase measurements: in the simplest case, one pair of images is assumed to contain the signature of topography only, while the other pair measures topography and change. Because the cross-track baseline of the two interferometric combinations are rarely the same, the sensitivity to topographic variation in the two generally differs. The phase differences in the topographic pair are scaled to match the frequency of variability in the topography-change pair. After scaling, the topographic phase differences are subtracted from the other, effectively removing the topography.

The first proof-of-concept experiment for spaceborne InSAR was conducted using SAR imagery obtained by the SeaSAT mission [13]. In the latter portion of that mission, the spacecraft was placed into a near-repeat orbit every 3 days. Gabriel et al. [14], using data obtained in an agricultural region in California, USA, detected surface elevation changes in some of the agricultural fields of the order of several cm over approximately 1 month. By comparing the areas with the detected surface elevation changes with irrigation records, they concluded that these areas were irrigated in between the observations, causing small elevation changes from increased soil moisture.

Fig. 5. A repeat track interferometer is similar to an along track interferometer. An aperture repeats its track and precisely measures motion of the surface between observations in the image phase difference. If the track does not repeat at exactly the same location, some topographic phase will also be present, which must be removed by the methods of differential interferometry to isolate the motion.

This differential interferometric SAR technique has since been applied to study minute terrain elevation changes caused by earthquakes and volcanoes. Several of the most important demonstrations will be described in a later section. A significant advantage of this remote sensing technique is that it provides a comprehensive view of the motion detected for the entire area affected. It is expected that this type of result will supplement ground-based measurements (e.g. GPS receivers), which are made at a limited number of locations. This technique has shown excellent promise to provide critical data for monitoring such natural hazards.

III. THEORY

A. Interferometry for Topographic Mapping

A.1 Basic Measurement Principles

The basic principles of interferometric radars are described in detail by a number of sources [8], [15], [16]. The following sections comprise the main results in the principles and theory of interferometry compiled from these and other papers, in a uniform notation and context. Appendix A describes aspects of SAR systems and image processing that are relevant to interferometry, including image compression, resolution, and pointing definitions.

A conventional SAR system resolves targets in the cross-track, or "range," direction by measuring the time it takes a radar pulse to propagate to the target and return to the radar. The azimuth location is determined from the Doppler frequency shift that results whenever the relative velocity between the radar and target is not zero. Geometrically, this is the intersection of a sphere centered at the antenna with radius equal to the radar range and a cone with generating axis along the velocity vector and cone angle proportional to the Doppler frequency as shown in Fig. 6. A target in the radar image could be located anywhere on the intersection locus which is a circle in the plane formed by the radar line of sight to the target and vector pointing from the aircraft to nadir. To obtain 3-dimensional position information, an additional measurement of elevation angle is needed. Interferometry using two or more SAR images provides a means of determining this angle.

Fig. 6. Target location in an IFSAR image is precisely determined by noting that the target location is the intersection of the range sphere, doppler cone and phase cone.

Interferometry can be understood conceptually by considering the signal return of elemental scatterers comprising each resolution element in a SAR image. Each resolution element of a SAR image can be represented as a complex phasor of the coherent backscatter from the scattering elements on the ground and the propagation phase delay, as illustrated in Fig. 7. The backscatter phase delay is the net phase of the coherent sum of the contributions from all elemental scatterers in the resolution element, each with their individual backscatter phases and their differential path delays relative to a reference surface normal to the radar look direction. Radar images observed from two nearby antenna locations have resolution elements with nearly the same complex phasor return, but with a different propagation phase delay. In interferometry, the complex phasor information of one image is multiplied by the complex conjugate phasor information of the second image to form an "interferogram," effectively canceling the common backscatter phase in each resolution element, but leaving a phase term proportional to the differential path delay. This is a geometric quantity directly related to the desired elevation angle.

Fig. 7. The interferometric phase difference is mostly due to the propagation delay difference. The (nearly) identical coherent phase from the different scatterers inside a resolution cell (mostly) cancels during interferogram formation.

For cross-track interferometers, two modes of data collection are commonly used: single transmitter, or historically "standard," mode where one antenna transmits and both interferometric antennas receive; and dual transmitter, or "ping-pong," mode where each antenna alternately transmits and receives its own echoes, as shown in Fig. 8. The measured phase differs by a factor of two depending on the mode as can be seen from Fig. 9.

Fig. 8. Illustration of standard vs. ping-pong mode of data collection. In standard mode, the radar transmits a signal out of one of the interferometric antennas only, and receives the echoes through both antennas A_1 and A_2 simultaneously. In "ping-pong" mode, the radar transmits alternatively out of the top and bottom antennas and receives the radar echo only through the same antenna.

In standard mode, the phase difference obtained in the interferogram is given by

$$\phi_{\text{npp}} = \frac{2\pi}{\lambda} \left(\rho_1 + \rho_1 - (\rho_2 + \rho_1) \right) = \frac{2\pi}{\lambda} (\rho_2 - \rho_1)$$
 (1)

where ρ_i is the range from antenna A_i to a point on the surface. In "ping-pong" mode, the phase is given by

$$\phi_{\rm pp} = \frac{4\pi}{\lambda} (\rho_2 - \rho_1) \tag{2}$$

One way to interpret this result is that the ping-pong operation effectively implements an interferometric baseline that is twice as long as that of the standard operation.

Fig. 9. SAR interferometry imaging geometry.

It is important to appreciate that only the principal values of the phase, modulo 2π , can be measured from the complex-valued resolution element. The total phase representing the range difference between the two observation points in general can be many multiples of the radar wavelength or, expressed in terms of phase, many multiples of 2π . The typical approach for determining the unique phase that is directly proportional to the range difference is to first determine to the relative phase between pixels via the so-called "phase-unwrapping" process. This connected phase field will then be adjusted by an overall constant multiple of 2π . The second step that determines the required multiple of 2π is referred to as "absolute phase determination." Figure 10 shows the principal value of the phase, the unwrapped phase and absolute phase for a pixel.

Fig. 10. Phase measurement in the interferogram is initially known modulo 2π . After unwrapping, relative phase measurements between all pixels in the interferogram are determined up to a constant multiple of 2π . Absolute phase determination finds the multiple of 2π that gives the correct proportionality to the range difference.

A.2 Interferometric Baseline and Height Reconstruction

In order to generate topographic maps or data for other geophysical applications using radar interferometry, we must relate the interferometric phase and other known or measurable parameters to the topographic height. It is also desirable to derive the sensitivity of the interferometrically determined topographic measurements to the interferometric phase and other known parameters. In addition, interferometric observations have certain geometric constraints that preclude valid observations for all possible image geometries. These issues are quantified below.

The interferometric phase as previously defined is proportional to the range difference from two antenna locations to a point on the surface. This range difference can be expressed in terms of the vector separating the two antenna locations, called the interferometric baseline. The range and azimuth position of the sensor associated with imaging a given scatterer depends on the portion of the synthetic aperture used to process the image (see Appendix A). Therefore the interferometric baseline depends on the processing parameters, and is defined as the difference between the location of the two antenna phase center vectors at the time when a given scatterer is imaged.

The equation relating the scatterer position vector, \vec{T} , a reference position for the platform \vec{P} , and the look vector, \vec{l} , is

$$\vec{T} = \vec{P} + \vec{l}
\vec{T} = \vec{P} + \rho \hat{l}$$
(3)

where ρ is the range to the scatterer and \hat{l} is the unit vector in the direction of \vec{l} . The position \vec{P} can be chosen arbitrarily, but is usually taken as the position of one of the interferometer antennas. Interferometric height reconstruction is the determination of a target's position vector from known platform ephemeris information, baseline information, and the interferometric phase. Assuming \vec{P} and ρ are known, interferometric height reconstruction amounts to the determination of the unit vector \hat{l} from the interferometric phase. Letting \vec{B} denote the baseline vector from antenna 1 to antenna 2, setting $\vec{P} = \vec{P_1}$ and defining

 $\vec{B} = \vec{P}_2 - \vec{P}_1 \quad B = |\vec{B}| \equiv \langle \vec{B}, \vec{B} \rangle^{1/2},$ (4)

we have the following expression for the interferometric phase

$$\phi = \frac{2\pi p}{\lambda} (\rho_2 - \rho_1) = \frac{2\pi p}{\lambda} (|\vec{l_2}| - |\vec{l_1}|)$$
 (5)

$$= \frac{2\pi p}{\lambda} \rho_1 \left[\left(1 - \frac{2\langle \hat{l}_1, \vec{B} \rangle}{\rho_1} + \left(\frac{B}{\rho_1} \right)^2 \right)^{1/2} - 1 \right]$$
 (6)

where p=2 for "ping-pong" mode systems and p=1 for standard mode systems. This expression can be simplified assuming $B \ll \rho$ by Taylor-expanding Eq. 6 to give

$$\phi \approx -\frac{2\pi p}{\lambda} \langle \hat{l}_1, \vec{B} \rangle \tag{7}$$

illustrating that the phase is approximately proportional to the projection of the baseline vector on the look direction, as illustrated in Fig. 11. This is the plane wave approximation of Zebker and Goldstein [8].

Fig. 11. When the plane wave approximation is valid the range difference is approximately the projection of the baseline vector onto a unit vector in the line of sight direction.

Specializing for the moment to the two dimensional case where the baseline lies entirely in the plane of the look vector and the nadir direction, $\vec{B} = (B\cos(\alpha), B\sin(\alpha))$ where α is the angle the baseline makes with respect to a reference horizontal plane. Then, Eq. 7 can be rewritten as

$$\phi = -\frac{2\pi p}{\lambda} B \sin(\theta - \alpha) \tag{8}$$

where θ is the look angle, the angle the line-of-sight vector makes with respect to nadir, shown in Fig. 9.

Fig. 12. (a) Radar brightness image of Mojave desert near Fort Irwin, California. (b) Interferogram of the area showing intrinsic fringe variability (c) Flattened interferogram assuming a reference surface at zero elevation above a spherical earth.

Figure 12b shows an interferogram of the Fort Irwin, California, generated using data collected on two consecutive days of the SIR-C mission. In this figure, the image brightness represents the radar backscatter and the color represents the interferometric phase, with one cycle of color equal to a phase change of 2π radians, or one "fringe." The rapid fringe variation in the cross track direction is mostly a result of the natural variation the line-of-sight vector across the scene. The fringe variation in the interferogram is "flattened" by subtracting the expected phase from a surface of constant elevation. The resulting fringes follow the natural topography more closely. Letting \hat{l}_0 be a unit vector pointing to a surface of constant elevation, h_0 , the flattened phase, ϕ_{flat} , is given by

$$\phi_{\text{flat}} = -\frac{2\pi p}{\lambda} \left(\langle \hat{l}, \vec{B} \rangle - \langle \hat{l}_0, \vec{B} \rangle \right) \tag{9}$$

where

$$\hat{l}_0 = (\sin \theta_0, -\cos \theta_0) \tag{10}$$

and $\cos \theta_0$ is given by the law of cosines

$$\cos \theta_0 = \frac{\rho_0^2 + (r_e + h_p)^2 - (r_e + h_0)^2}{2(r_e + h_p)\rho_0}$$
(11)

assuming a spherical Earth with radius r_e and a slant range to the reference surface ρ_0 . The flattened Fort Irwin inteferogram is shown in Fig. 12c where the fringes more closely mimic the topographic contours of a conventional map.

The intrinsic fringe frequency in the slant plane interferogram is given by

$$\frac{\partial \phi}{\partial \rho} = \frac{\partial \phi}{\partial \theta} \frac{\partial \theta}{\partial \rho} \tag{12}$$

$$= \frac{2\pi p}{\lambda \rho} B \cos(\theta - \alpha) \,. \tag{13}$$

$$\frac{1}{\rho \sin \theta} \left[-\frac{\rho}{h_p + r_e} + \cos \theta + \frac{h_0 + r_e}{h_p + r_e} \frac{\sin \tau_c}{\sin(i - \tau_c)} \right] \tag{14}$$

where

$$\sin i = \frac{h_p + r_e}{h_0 + r_e} \sin \theta \tag{15}$$

and i is the local incidence angle relative a spherical surface, h_p is the height of the platform, and τ_c is the surface slope angle in the cross track direction as defined in Fig. 9. From Eq. 14, it is seen that fringe frequency is proportional to the perpendicular component of the baseline, defined as

$$B_{\perp} \equiv B\cos(\theta - \alpha) \tag{16}$$

As B_{\perp} increases or as the local terrain slope approaches the look angle, the fringe frequency increases. Slope dependence of the fringe frequency can be observed in Fig. 12c where the fringe frequency increases on slopes facing toward the radar and is less on slopes facing away from the radar. It is also evident from Eq. 14 that the fringe frequency is inversely proportional to λ , thus longer wavelengths result in smaller fringe frequencies. If the phase changes by 2π or more across the range resolution element, $\Delta \rho$, the different contributions within the resolution cell do not add to a well defined phase resulting in what is commonly referred to as decorrelation of the interferometric signal. Thus, in interferometry, an important parameter is the critical baseline, defined as the perpendicular baseline at which the phase rate reaches 2π per range resolution element. From Eq. 14, the critical baseline satisfies the proportionality relationship

$$B_{\perp, \rm crit} \propto \frac{\lambda}{\Delta \rho}$$
. (17)

This is a fundamental constraint for interferometric radar systems. Also, the difficulty in phase unwrapping increases (see Sec. III-D.1) as the fringe frequency approaches this critical value.

Full three dimensional height reconstruction is based on the observation that the target location is the intersection locus of three surfaces: the range sphere, Doppler cone, and phase cone described earlier. The cone angles are defined relative to the generating axes determined by the velocity vector for the Doppler cone and the baseline vector for the phase cone. (Actually the phase surface is a hyperboloid, however for most applications where the plane wave approximation is valid, the hyperboloid degenerates to a cone.) The intersection locus is the solution of the system equations

$$\rho = |\vec{P} - \vec{T}| \quad \text{Range Sphere}$$

$$f = 2\langle \vec{v}, \hat{l} \rangle / \lambda \quad \text{Doppler Cone}$$

$$\phi = -2\pi p \langle \vec{B}, \hat{l} \rangle / \lambda \quad \text{Phase Cone}$$
(18)

which can be solved for \hat{l} . These equations appear to be non-linear, but by choosing an appropriate local coordinate basis, they can be readily solved for \hat{l} . To illustrate, let $\vec{P} = (y_0, h_p)$ be the platform position vector. Then from the basic height reconstruction equation Eq. 3

$$\vec{T} = (y_0, h_p) + \rho \left(\sin \theta, -\cos \theta\right). \tag{19}$$

To relate the look angle to the inteferometric phase, Eq. 8 can be inverted for θ yielding

$$\theta = \sin^{-1}\left(\frac{-\lambda\phi}{2\pi pB}\right) + \alpha \tag{20}$$

It is immediate from the above expressions that reconstruction of the scatterer position vector depends on knowledge of the platform location, the interferometric baseline length, orientation angle and the interferometric phase. To generate accurate topographic maps, radar interferometry places stringent requirements on knowledge of the platform and baseline vectors. In the above discussion, atmospheric effects are neglected. Appendix B develops the correction for atmospheric refraction in an exponential, horizontally stratified atmosphere, showing that the bulk of the correction can be made by altering the effective speed of light through the refractive atmosphere. Refractivity fluctuation due to turbulence in the atmosphere is a minor effect [17].

To illustrate these theoretical concepts in a more concrete way, we show in Fig. 13 a block diagram of the major steps in the processing data for topographic mapping applications, from raw data collection to generation of a digital topographic model. The description assumes simultaneous collection of the two interferometric channels; however, with minor modification, the procedure outlined applies to repeat track processing as well.

Fig. 13. Block diagram showing the major steps in interferometric processing to generate topographic maps. Data for each interferometric channel are processed to full resolution images using the platform motion information to compensate the data for perturbations from a straight line path. One of the complex images is resampled to overlay the other, and an interferogram is formed by cross-multiplying images, one of which is conjugated. The resulting interferogram is averaged to reduce noise. Then, the principal value of the phase for each complex sample is computed. To generate a continuous height map, the two-dimensional phase field must be unwrapped. After the unwrapping process, an absolute phase constant is determined. Subsequently, the three dimensional target location is performed with corrections applied to account for tropospheric effects. A relief map is generated in a natural coordinate system aligned with the flight path. Gridded products may include the target heights, the SAR image, a correlation map, and a height error map.

B. Interferometry for Motion Mapping

The theory described above assumed that the imaged surface is stationary over time, or that the surface is imaged by the interferometer at a single instant. When there is motion of the surface between radar observations there is an additional contribution to the interferometric phase variation. Figure 14 shows the geometry when a surface displacement occurs between the observation at \vec{P}_1 (at time t_1) and the observation at \vec{P}_2 (at $t_2 > t_1$). In this case, \vec{l}_2 becomes

$$\vec{l}_2 = \vec{T} + \vec{D} - \vec{P}_2 = \vec{l}_1 + \vec{D} - \vec{B} \tag{21}$$

where \vec{D} is the displacement vector of the surface from t_1 to t_2 . The interferometric phase expressed in terms of this new vector is

$$\phi = \frac{4\pi}{\lambda} \left(\langle \vec{l}_1 + \vec{D} - \vec{B}, \vec{l}_1 + \vec{D} - \vec{B} \rangle^{1/2} - \rho_1 \right)$$
 (22)

Assuming as above that $|\vec{B}|, |\vec{D}|$, and $|\langle \vec{B}, \vec{D} \rangle|$ are all much smaller than ρ_1 , the phase reduces to

$$\phi = \frac{4\pi}{\lambda} \left(-\langle \vec{l}_1, \vec{B} \rangle + \langle \vec{l}_1, \vec{D} \rangle \right) \tag{23}$$

This justifies the usual formulation in the literature that

$$\phi_{\text{obs}} = \phi_{\text{topography}} + \phi_{\text{displacement}}.$$
 (24)

In some applications, the displacement phase represents a nearly instantaneous translation of the surface resolution elements, e.g. earthquake deformation. In other cases, such as glacier motion, the displacement phase represents a steady velocity tracked over the time between observations. Intermediate cases include slow and/or variable surface motions, such as volcanic inflation or surgling glaciers.

Fig. 14. Geometry of displacement interferometry. Surface element has moved in a coherent fashion between observation A_1 made at time t_1 and observation A_2 made at time t_2 . The displacement can be of any sort - continuous or instantaneous, steady or variable - but the detailed scatterer arrangement must be preserved in the interval for coherent observation.

Equations 23 and 24 highlight that the interferometer measures the projection of the displacement vector in the radar line-of-sight direction. To reconstruct the vector displacement, observations must be made from different aspect angles.

The topographic phase term is not of interest for displacement mapping, and must be removed. Several techniques have been developed to do this. They all essentially derive the topographic phase from another data source, either a digital elevation model (DEM) or another set of interferometric data. The selection of a particular method for topography measurement depends heavily on the nature of the motion (steady or episodic), the imaging geometry (baselines and time separations), and the availability of data.

It is important to appreciate the increased precision of the interferometric displacement measurement relative to topographic mapping precision. Consider a discrete displacement event such as an earthquake where the surface moves by a fixed amount \vec{D} in a short time period. Neither a pair of observations acquired before the event (pair "a"), nor a pair after the event (pair "b") would measure the displacement directly, but would measure it through the change in topography. According to Eq. 22, and assuming the same imaging geometry for "a" and "b" without loss of generality, the phase difference between these two interferograms (that is the difference of phase differences) is

$$\phi = \phi_{a} - \phi_{b} \tag{25}$$

$$= \frac{4\pi}{\lambda} \left[\left(\langle \vec{l}_{1} - \vec{B}, \vec{l}_{1} - \vec{B} \rangle^{1/2} - \rho_{1} \right) - \left(\langle \vec{l}_{1} + \vec{D} - \vec{B}, \vec{l}_{1} + \vec{D} - \vec{B} \rangle^{1/2} - \langle \vec{l}_{1} + \vec{D}, \vec{l}_{1} + \vec{D} \rangle^{1/2} \right) \right]$$

$$= 0 \tag{27}$$

to first order, because \vec{D} appears in both the expression for $\vec{l_2}$ and $\vec{l_1}$. The nature of the sensitivity difference inherent between Eqs. 23 and 27 can be seen in the "flattened" phase (see Eq. 9) of an interferogram, often written [18]

$$\phi = -\frac{4\pi}{\lambda}B\cos(\theta_0 - \alpha)\frac{z}{\rho_0\sin\theta_0} + \frac{4\pi}{\lambda}\delta\rho \tag{28}$$

where $\delta\rho$ is the surface displacement between imaging times in the radar line of sight direction, and z is the topographic height above the reference surface. In this formulation, the phase difference is far more sensitive to *changes* in topography (surface displacement) than to the topography itself. From Eq. 28, $\delta\rho = \lambda/2$ gives one cycle of phase difference, while z must change by a substantial amount, essentially ρ_0/B , to affect the same phase change.

The time interval over which the displacement is measured must be matched to the geophysical signal of interest. For ocean currents the temporal baseline must be of the order of a fraction of a second to achieve interferometric correlation. At the other extreme, temporal baselines of several years may be required to make accurate measurements of slow deformation processes such as interseismic strain.

C. Sensitivity Equations and Accuracy

C.1 Sensitivity Equations and Error Model

In design tradeoff studies of IFSAR systems, it is often convenient to know how interferometric performance varies with system parameters and noise characteristics. Sensitivity equations are derived by differentiating the basic interferometric height reconstruction equation Eq. 3 with respect to the parameters needed to determine \vec{P} , \hat{l} and ρ . Dependency of the quantities in the equation on the parameters typically measured by an interferometric radar is shown in Fig. 15. The sensitivity equations may be extended to include additional dependencies such as position and baseline metrology system parameters as needed for understanding a specific system's performance or for interferometric system calibration.

Fig. 15. Sensitivity tree for target location using interferometric height reconstruction.

To develop a basic set of sensitivity equations, we use the standard broadside geometry where the baseline is in a plane perpendicular to the velocity vector. With this geometry the baseline and velocity vectors are given by

$$\vec{B} = (0, B\cos\alpha, B\sin\alpha) \tag{29}$$

$$\vec{v} = (v, 0, 0) \tag{30}$$

where B is baseline length, α the baseline orientation angle, θ is the look angle, as shown in Fig. 16.

Fig. 16. Baseline and look angle geometry as used in sensitivity formulas.

The sensitivity equations can be stated in vector form [17], assuming that $B \ll \rho$, as

$$\Delta \vec{T} = \Delta \vec{P} + \Delta \rho \hat{l} + \frac{\rho}{\langle \hat{B}, \hat{l} \times \hat{v} \rangle}$$

$$\left[\left(\frac{-\lambda \Delta \phi}{2\pi B} - \langle \vec{l}, \frac{\Delta \vec{B}}{B} \rangle \right) \hat{l} \times \hat{v} + \langle \hat{l}, \frac{\Delta \vec{v}}{v} \rangle \hat{l} \times \hat{B}$$

$$- \left(\hat{v} \times \hat{B} - \langle \hat{B}, \hat{l} \times \hat{v} \rangle \hat{l} \right) \frac{\Delta \lambda}{\lambda} \right]$$
(31)

Observe from Eq. 31 that interferometric position determination error is directly proportional to platform position error, range errors lie on a vector parallel to the line of sight, \hat{l} , baseline and phase errors result in position errors which lie on a vector parallel to $\hat{l} \times \hat{v}$, and velocity errors result in position errors on a vector parallel to $\hat{l} \times \hat{B}$. Since the look vector \hat{l} in an interferometric mapping system has components both parallel and perpendicular to nadir, baseline and phase errors contribute simultaneously to planimetric and height errors. For broadside mapping geometries note that velocity errors do not not contribute to target position errors.

It is often useful to have explicit expressions for the various error sources in terms of the standard interferometric system parameters and these are found in the equations below. Differentiating Eq. 3 with respect to an arbitrary parameter ξ yields

$$\frac{\partial \vec{T}}{\partial \xi} = \frac{\partial \vec{P}}{\partial \xi} + \rho \frac{\partial \hat{l}}{\partial \xi} + \frac{\partial \rho}{\partial \xi} \hat{l}$$
(32)

The parameters we substitute for ξ are the interferometric phase ϕ , baseline length B, baseline orientation angle α , range ρ , and position \vec{P} . To highlight the essential features of the interferometric sensitivity, we simplify the geometry to a flat earth and broadside imaging case. These formulas are useful for assessing system performance or making trade studies. The full vector equation is needed for use in system calibration.

The sensitivity of the target position to platform position in the along track direction s, cross track direction c, and vertical direction h is given by

$$\frac{\partial \vec{T}}{\partial P_s} = \begin{bmatrix} 1\\0\\0 \end{bmatrix}, \qquad \frac{\partial \vec{T}}{\partial P_c} = \begin{bmatrix} 0\\1\\0 \end{bmatrix}, \qquad \frac{\partial \vec{T}}{\partial P_h} = \begin{bmatrix} 0\\0\\1 \end{bmatrix}$$
(33)

Note that an error in the aircraft position merely translates the reconstructed position vector in the direction of the platform position error. Only platform position errors exhibit complete independence of target location within the scene.

The sensitivity of the target position to range errors is given by

$$\frac{\partial \vec{T}}{\partial \rho} = \begin{bmatrix} 0 \\ \sin \theta \\ -\cos \theta \end{bmatrix} = \hat{l}_{sch} \tag{34}$$

Note that range errors occur in the direction of the line-of-sight vector. Targets with small look angles have larger vertical than horizontal errors, whereas targets with look angles greater than 45° have larger cross track position errors than vertical errors.

The sensitivity of the target position to errors in the baseline length, and baseline roll angles are given by

$$\frac{\partial \vec{T}}{\partial B} = \frac{\rho}{B} \tan(\theta - \alpha) \begin{bmatrix} 0 \\ -\cos\theta \\ -\sin\theta \end{bmatrix}$$
 (35)

and

$$\frac{\partial \vec{T}}{\partial \alpha} = \rho \begin{bmatrix} 0 \\ \cos \theta \\ \sin \theta \end{bmatrix},\tag{36}$$

Sensitivity to the interferometric phase is given by

$$\frac{\partial \vec{T}}{\partial \phi} = \left(\frac{-\lambda \rho}{2p\pi B \cos(\theta - \alpha)}\right) \begin{bmatrix} 0\\ \cos \theta\\ \sin \theta \end{bmatrix}$$
 (37)

where p is 1 or 2 for single transmit or ping-pong modes.

A parameter often used in interferometric system analysis and characterization is the ambiguity height, the amount of height change that leads to a 2π change in interferometric phase. The ambiguity height, h_a , is given by

$$h_a = 2\pi \frac{\partial h}{\partial \phi} = \frac{\lambda \rho \sin \theta}{pB \cos(\theta - \alpha)}$$
 (38)

Table III-C.1 shows predicted interferometric height error sensitivities for the C-band TOPSAR [19] and Shuttle Radar Topography Mission (SRTM) [20] radar systems. Although these systems have different mapping resolutions, imaging geometries, and map accuracy requirements, there are some key similarities between these systems. Both of these systems require extremely precise knowledge of

TABLE III-C.1: SENSITIVITY FOR TWO INTERFEROMETRIC SYSTEMS

Interferometric Parameter	TOPSAR			SRTM		
	Parameter	· C	h	Parameter	(,	h
	Error	Error	Error	Error	Error	Error
Baseline Length	1.0 mm	0.2 m	0.3 m	3 mm	2.5 m	4.4 m
Baseline Angle	$15.0 \ \mathrm{arcsec}$	0.9 m	1.5 m	9 arcsec	9.8 m	$15.3 \mathrm{m}$
Random Phase	3.0 mm	0.8 m	1.4 m	8°	7.1 m	10.9 m
Range	$0.5 \mathrm{m}$	0.8 m	0.5 m	3 m	2.5 m	1.6 m
Platform Position	1.0 m	1.0 m	1.0 m	1 m	1.0 m	1.0 m

the baseline length and orientation angle - millimeter or better measurements of the baseline length and 10's of arc second angle measurements for the baseline orientation angle. These requirements are typical of most InSAR systems, and generally necessitate either an extremely rigid and controlled baseline, a precise baseline metrology system, or both.

Phase accuracy requirements for interferometric systems typically range from 0.1 - 10 degrees. This imposes rather strict monitoring of phase changes not related to the imaging geometry in order to produce accourate topographic maps. Both the TOPSAR and SRTM system use a specially-designed calibration signal to remove radar electronically-induced phase delays between the interferometric channels.

D. The Phase in Interferometry

D.1 Phase Unwrapping

The phase of the interferogram must be unwrapped to remove the modulo- 2π ambiguity before estimating topography or surface displacement. There are two main approaches to phase unwrapping. The first class of algorithms is based on the integration with branch cuts approach initially developed by Goldstein et al. [13]. A second class of algorithms is based on a least-squares (LS) fitting of the unwrapped solution to the gradients of the wrapped phase. The initial application of least-squares method to interferometric phase unwrapping was by Ghiglia and Romero [22], [23]. Fornaro et al. [24] have derived another method based on a Green's function formulation, which has been shown to be theoretically equivalent to the LS method [25]. Other unwrapping algorithms that do not fall into either of these categories have been introduced [26], [27], [28], [29], [30], [31], and several hybrid algorithms and new insights have arisen [32], [33], [34], [35], [36].

Branch-cut methods

A simple approach to phase unwrapping would be to form the first differences of the phase at each image point in either image dimension, and then integrate the result. Direct application of this approach, however, allows local errors due to phase noise to propagate, causing errors across the full SAR scene[13]. Branch-cut algorithms attempt to isolate sources of error prior to integration. The basic idea is to unwrap the phase by choosing only paths of integration that lead to self-consistent solutions [13]. The first step is to difference the phase so that differences are mapped into the interval $(-\pi,\pi]$. In performing this operation, it is assumed that the true (unwrapped) phase does not change by more than $\pm \pi$ between adjacent pixels. When this assumption is violated, either from statistical phase variations or rapid changes in the true intrinsic phase, inconsistencies are introduced that can lead to unwrapping errors.

The unwrapped solution should, to within a constant of integration, be independent of the path of integration. This implies that in the error-free case, the integral of the differenced phase about a closed path is zero. Phase inconsistencies are therefore indicated by non-zero results when the phase

difference is summed around the closed paths formed by each mutually neighboring set of four pixels. These points, referred to as "residues" in the literature, are classified as either positively or negatively "charged," depending on the sign of the sum (the sum is by convention performed in clockwise paths). Figure 17 illustrates how a local error at a residue propagates. Two possible paths of integration from points A to B are shown. If no residue were present, the integral from A to B along either path would yield the same result. With a residue, however, the results of integration along either path differ, yielding inconsistent solutions at B. This is true no matter how far the point B is from the residue, so that the local error at the residue causes error in the overall interferometric result.

Fig. 17. An example of two possible paths of integration from A to B with a nearby residue.

Because integration of the differenced phase about a closed path yields a value equal to the sum of the enclosed residues, paths of integration that encircle a net charge must be avoided. This is accomplished by connecting oppositely charged residues with branch cuts, which are lines the path of integration cannot cross. Figure 18 shows an example of a branch cut. As the figure illustrates, it is not possible to choose a path of integration that does not cross the cut, yet contains only a single residue. An interferogram may have a slight net charge, in which case the excess charge can be "neutralized" with a connection to the border of the interferogram. Once branch cuts have been selected, phase unwrapping is completed by integrating the differenced phase subject to the rule that paths of integration do not cross branch cuts.

Fig. 18. An example of a branch cut and allowable and forbidden paths of integration.

Fig. 19. Cut dependencies of unwrapped phase. a) shortest path cuts b) better choice of cuts.

The method for selection branch cuts is the most difficult part of the design of any branch-cut based unwrapping algorithm and is the key distinguishing feature of members of this class of algorithms. In most cases the number of residues is such that evaluating the results of all possible solutions is computationally intractable. Thus, branch cut selection algorithms typically employ heuristic methods to limit the search space to a reasonable number of potentially viable solutions [13], [32], [37].

Figure 19 shows a schematic example of a phase discontinuity and how different choices of cuts can affect the final result. In Fig. 19a, the shortest possible set of branch cuts is used to connect the residues. This choice of branch cuts forces the path of integration to cross a region of true phase shear, causing the phase in the shaded region to be unwrapped incorrectly and the discontinuity to be inaccurately located across the long vertical branch cut. Figure 19b shows a better set of branch cuts where the path of integration is restricted from crossing the phase shear. With these cuts, the phase is unwrapped correctly for the shaded region and the discontinuity across the branch cut closely matches the true discontinuity.

A commonly cited misconception regarding branch-cut algorithms is that operator intervention is needed to succeed [23][24]. Fully automated branch cuts algorithms have been used to select branch cuts for a wide variety of interferometric data from both airborne and spaceborne sensors.

Least-squares methods

An alternate set of phase unwrapping methods is based on a least squares approach. These algorithms minimize the difference between the gradients of the solution and the wrapped phase in a

least-squares sense. Following the derivation of Ghiglia and Romero [23] the sum to be minimized is

$$\sum_{i=0}^{M-2} \sum_{j=0}^{N-1} \left(\phi_{i,j} - \phi_{i-1,j} - \Delta_{i,j}^{x} \right)^{2} + \sum_{i=0}^{M-1} \sum_{j=0}^{N-2} \left(\phi_{i,j} - \phi_{i,j-1} - \Delta_{i,j}^{y} \right)^{2}$$
(39)

where $\phi_{i,j}$ is the unwrapped solution corresponding to the wrapped values $\psi_{i,j}$ and

$$\Delta_{i,j}^x = W(\psi_{i,j} - \psi_{i-1,j}) \tag{40}$$

and

$$\Delta_{i,j}^{y} = W(\psi_{i,j} - \psi_{i,j-1}) \tag{41}$$

with the operator W() wrapping values into the range $(-\pi,\pi]$ with the appropriate addition of $\pm 2\pi$. In this equation, N and M are the image dimensions.

The summation in Eq. 39 can be reworked so that for each set of indices i, j,

$$\phi_{i+1,j} + \phi_{i-1,j} + \phi_{i,j+1} + \phi_{i,j-1} = \rho_{i,j}, \tag{42}$$

where

$$\rho_{i,j} = (\Delta_{i,j}^x - \Delta_{i-1,j}^x) - (\Delta_{i,j}^y - \Delta_{i,j-1}^x)$$
(43)

This equation represents a discretized version of Poisson's equation. The least-squares problem then may be formulated as the solution of a linear set of equations

$$A\phi = \rho \tag{44}$$

where A is an NM by NM sparse matrix and the vectors ϕ and ρ contain the phase values on the left and right hand sides of Eq. 43, respectively. For typical image dimensions, the matrix A is too large to obtain a solution by direct matrix inversion. A computationally fast and efficient solution, however, can be obtained using an FFT based algorithm [23].

The unweighted LS solution is sensitive to inconsistencies in the wrapped phase (i.e., residues), leading to significant errors in the unwrapped phase. A potentially more robust approach is to use a weighted LS solution. In this case, an iterative computational scheme (based on the FFT algorithm) is necessary to solve Eq. 44, leading to significant increases in computation time. Other computational techniques have been used to further improve throughput performance [33], [34]

Branch-cut vs. least-squares methods

The performance of least-squares and branch-cut algorithms differ in several important ways. Branch-cut algorithms tend to "wall-off" areas with high residue density (for example a lake in a repeat-pass interferogram where the correlation is zero) so that holes exist in the unwrapped solution. In contrast, LS algorithms provide continuous solutions even where the phase noise is high. This can be considered both a strength and a weakness of the LS approach since on one hand LS leaves no holes but on the other hand it may provide erroneous data in these areas.

Errors in branch cut solution are always integer multiples of 2π (i.e., when the unwrapped solution is rewrapped it equals the original wrapped phase). These errors are localized in the sense that the result consists of two types of regions: those that are unwrapped correctly and those that have error that is an integer multiple of 2π . In contrast, LS algorithms yield errors that are continuous and distributed over the entire solution. Long wavelength errors can be introduced during LS unwrapping. For example, unweighted LS squares solutions have been shown to be biased estimators of slope [35]. Whether slope biases are introduced for weighted LS depends on the particular implementation of the weighting scheme.

Phase unwrapping using branch cuts is a well established and mature method for interferometric phase unwrapping. It has been applied to a huge volume of interferometric data and will be used as

the algorithm for the Shuttle Radar Topography Mission data processor (see below). Unweighted LS algorithms are not sufficiently robust for most practical applications [23], [33]. While weighted LS can yield improved results, the results are highly dependent on the selection of weighting coefficients. The selection of these weights is a problem of similar complexity to that of selecting branch cuts. Automated weighting selection algorithms and weighted LS algorithms have not yet been applied widely or validated using a diverse interferometric data set. Thus, further development is needed to fully establish the capability of this class of algorithms.

D.2 Absolute Phase

Successful phase unwrapping will establish the correct phase differences between neighboring pixels. The phase value required to make a geophysical measurement is that which is proportional to range delay. This phase is called the "absolute phase." Usually the unwrapped phase will differ from the absolute phase by an integer multiple of 2π (and possibly a calibration phase factor which we will ignore here). Assuming that the phases are unwrapped correctly, this integer is a single constant throughout a given interferometric image set. There are a number of ways to determine the absolute phase. In topographic mapping situations the elevation of a reference point in the scene might be known and given the mapping geometry, including the baseline, one can calculate the absolute phase, e.g. from Eq. 19. However, in the absence of any reference, it may be desirable to determine the absolute phase from the radar data.

To characterize the absolute phase, consider the transmitted signal in channel i (=1 or 2) given by:

$$p_i(t) = h_i(t) \exp(j2\pi f_0 t) \tag{45}$$

After down-conversion to baseband, the received echo from a target is

$$s_i(t) = h_i(t - t_{di}) \exp(-j2\pi f_0 t_{di}), \tag{46}$$

delayed by time delay t_{di} .

When shifting the two channels to coregister them, the baseband signal is shifted and phase corrected, equivalent to shifting the signal at the carrier frequency. To achieve the time coregistration each channel is shifted by $t_{di,REF} - t_{d0,REF}$ (one of which could be arbitrarily set to zero), the delay difference between track i and the reference track 0, given a target in a reference elevation plane. The signal after the range shift and phase rotation is:

$$g_{i}(t) = s_{i}(t + t_{di,REF} - t_{d0,REF}) \exp(j2\pi f_{0}(t_{di,REF} - t_{d0,REF}))$$

$$= h_{i}(t + t_{di,REF} - t_{d0,REF} - t_{di}) \exp(j2\pi f_{0}(t_{di,REF} - t_{d0,REF} - t_{di}))$$
(47)

Assuming identical transfer functions h_0 for the two channels, the interferometric correlation function, or interferogram, is:

$$c(t, f_0) = g_i(t)g_j^*(t)$$

$$= h_0(t - t_{di} + t_{di,REF} - t_{d0,REF})h_0^*(t - t_{dj} + t_{dj,REF} - t_{d0,REF})$$

$$= \exp(-j2\pi f_0(t_{di} - t_{dj} - (t_{di,REF} - t_{dj,REF})))$$
(48)

The interferogram phase is proportional to the carrier frequency and the difference between the actual time delay differences and that assumed during the co-registration step. Two methods proposed to determine the absolute phase automatically, without using target information [38], [39], exploit these relationships.

The "split-spectrum" estimation algorithm, divides the available RF-bandwidth in two or more separate subbands. A differential interferogram formed from two subbanded interferograms, with carrier frequencies f_+ and f_- , has the phase

$$\phi_{\pm}(t) = -2\pi (f_{+} - f_{-})(t_{di} - t_{dj} - (t_{di,REF} - t_{dj,REF}))$$
(49)

This shows that the phase of the differential interferogram is equivalent to that of an interferogram with a carrier which is the difference of the carrier frequencies of the two interferograms used. The difference $f_+ - f_-$ should be chosen such that the differential phase is always in the range $|-\pi,\pi|$, making the differential phase unambiguous. Thus, from the phase term in Eq. 48 and Eq. 49, a relationship between the original and differential interferometric phase is established:

$$\phi_0(t) = \phi_{\pm}(t) \frac{f_0}{f_+ - f_-} \tag{50}$$

The noise in the differential interferogram is comparable to that of the "standard" interferogram, but typically larger by a factor of two. After scaling the differential absolute phase value, the noise at the actual RF carrier phase is typically much larger than 2π . Instead, we can use that after phase unwrapping,

$$\phi_{\text{unw}}(t) = \phi_0(t) - n2\pi \tag{51}$$

which leads us to an estimator for the integer multiple of

$$n = \frac{1}{2\pi} \left(\phi_{\pm}(t) \frac{f_0}{f_+ - f_-} - \phi_{\text{unw}}(t) \right)$$
 (52)

This estimate can be averaged over all points in the interferogram allowing significant noise reduction. The "residual delay" estimation technique, is based on the observation that the absolute phase is proportional to the signal delay. The basis of SAR interferometry is that the phase measurement is the most accurate measure of delay. The signal delay measured directly from the full signal (e.g. by correlation analysis or modified versions thereof) is an unambiguous determination of the delay, but to determine the channel-to-channel delay accurately, a large correlation basis is required. For such a large estimation area, however, the inherent channel signal delay difference is seldom constant because of parallax effects, and so delay estimates from direct image correlation can rarely attain the required accuracy.

The unwrapped phase can be used to mitigate this problem. As the unwrapped phase is an estimate of the channel to channel delay difference, the unwrapped phase is a measure of the spatially varying delay shift required to interpolate one image to have the same delay as the other channel. If the unwrapped phase is identical to the absolute phase, the two image delays will be identical (except for noise) after the interpolation. If on the other hand the unwrapped and the absolute phases differ by an integer number of 2π then the delay difference between the two channels will be offset by this same integer number of RF-cycles. This delay is constant throughout the image, and can thus be estimated over large image areas.

From Eqs. 48 and 51, we have:

$$\phi_{\text{unw}}(t) = -2\pi f_0(t_{di} - t_{dj} - (t_{di,REF} - t_{dj,REF}) + \frac{n}{f_0})$$
 (53)

where n is unknown. Using $\phi_{unw}(t)$ we can resample and phase shift channel j:

$$g_{j}^{S}(t) = g_{j}(t - \frac{\phi_{\text{unw}}}{-2\pi f_{0}}) \exp(j\phi_{\text{unw}})$$

$$= h_{j}(t - t_{di} + t_{di,\text{REF}} - t_{d0,\text{REF}} - \frac{n}{f_{0}}) \exp(j2\pi f_{0}(-t_{di} + t_{di,\text{REF}} - t_{d0,\text{REF}}))$$
(54)

Thus if $h_i(t) = h_j(t)$ then $g_j^S(t) = g_i(t - n/f_0)$. For a given data set, after resampling and phase shifting one of the complex images, the two images will be identical with the exception of a time delay difference (= two times the range shift divided by the speed of light) which is: 1) constant over the

image processed; and 2) proportional to n, the number of cycles by which the unwrapped phase differs from the absolute phase. The residual integer multiple of 2π can thus be estimated from precision delay estimation methods.

For this procedure to work the channel delay difference must be measured to hundredths or even thousandths of a pixel in range (significantly better than the ratio of λ/p to the resolution cell size) and very accurate algorithms for both interpolation and delay estimation are required. Even small errors are of concern. Thermal noise is one error source, but due to its zero mean character, it is generally not the key limitation [40]. Systematic errors are of much larger concern. For example, if the interpolations in the SAR processor are not implemented carefully, they will modify the transfer functions and introduce systematic errors in the absolute phase estimate. In the case of the splitspectrum estimation even small transfer function changes will have a significant impact on the absolute phase due to the very large multiplier involved $(f_0/(f_+-f_-))$. In the case of the residual delay estimator even small changes in the system impulse response function will bias the correlation, a critical concern when accuracies on the order of a thousandth of a pixel are needed. System transfer functions are also critical, as indicated when discussing interpolations. Ideally the transfer functions $h_{01}(t)$ and $h_{02}(t)$ should be identical, real, and symmetrical. However, when the transfer functions of the two channels are different and furthermore varying across the swath, it can be very difficult to estimate the absolute phase accurately. A particularly troubling error source is multipath on the sensor as it will cause phase and impulse response errors which are varying over the swath [41].

E. Interferometric Correlation and Phenomenology

The discussion in the preceding sections implicitly assumed that the interferometric return could be regarded as being due to a point scatterer. For most situations, this will not be the case: scattering from natural terrain is generally considered as the coherent sum of returns from many individual scatterers within any given resolution cell. This is relevant in cases where the surface is rough compared to the radar wavelength. This coherent addition of returns from many scatterers gives rise to "speckle" [42]: the scattered field at the receiver plane oscillates rapidly in space in both phase and amplitude. For cases where there are many scatterers, the coherent summation of the scatterers' responses will obey circular-Gaussian statistics [42]. The relationship between the scattered fields at the interferometric receivers is then determined by the statistics at each individual receiver, and by the complex correlation function, γ , defined as

$$\gamma = \frac{\langle E_1 E_2^* \rangle}{\sqrt{\langle |E_1|^2 \rangle \langle |E_2|^2 \rangle}} \tag{55}$$

where E_i represents the SAR return at the *i* antenna, and angular brackets denote averaging over the ensemble of speckle realizations. For completely coherent scatterers such as point scatterers, we have that $\gamma = 1$, while $\gamma = 0$ when the scattered fields at the antennas are independent.

The effect of field decorrelation is the apparent increase in noise of the estimated interferometric phase. The actual dependence of the phase variance on the correlation and the number of independent estimates used to derive the phase was characterized by Monte Carlo simulation [9]. Rodríguez and Martin [15] presented the analytic expression for the phase variance σ_{ϕ}^2

$$\sigma_{\phi}^{2} = \frac{1}{2N_{L}} \frac{1 - \gamma^{2}}{\gamma^{2}} \tag{56}$$

starting from the Cramer-Rao bound [43]. This limit is approached as the number of looks increases, and is a reasonable approximation when the number of looks is greater than four. An exact expression for the phase variance can be obtained starting from the probability density function for the phase when $N_L = 1$, and then extended for an arbitrary number of looks [42]. The expression, however, is quite complicated and must be evaluated numerically in practice.

In addition to speckle decorrelation, thermal noise in the interferometric channels also introduces phase noise in the interferometric measurement. Since the noise is also circular-Gaussian and independent in each channel, one can show that γ_N , the correlation due to thermal noise alone, can be written as

$$\gamma_N = \frac{1}{\sqrt{1 + \text{SNR}_1^{-1}} \sqrt{1 + \text{SNR}_2^{-1}}}$$
 (57)

where SNR_i denotes the signal-to-noise ratio for the *i* channel. In addition to thermal noise, which is additive, SAR returns also have other noise components, due to, for example, range and Doppler ambiguities. An expression for the decorrelation due to this source of error can only be obtained for homogeneous scenes, since, in general, the noise contribution is scene dependent. Typically for simplicity these ambiguities are treated as additive noise as part of the overall system noise floor.

The decorrelation due to speckle can be understood in terms of the van Cittert-Zernike (vCZ) theorem [42]. In its traditional form, the vCZ theorem states that the correlation function of the field due to scatterers located on a plane perpendicular to the look direction is proportional to the Fourier transform of the scatterer intensity, provided the scatterers can be regarded as independent from point to point.

The vCZ theorem was extended to the IFSAR geometry [9], and was subsequently expanded to include volume scattering [15], and to include arbitrary point target responses [44]. Further contributions [45] showed that part of the decorrelation effect could be removed if slightly different radar frequencies were used for each interferometric channel, so that the component of the incident wavenumbers projected on the scatterer plane from both antennas is identical.

Physically, the speckle decorrelation is due to the fact that, after removing the phase contribution from the center of the resolution cell, the phases from the scatterers located away from the center are slightly different at each antenna (see Fig. 7). The degree of decorrelation can then be estimated from the differential phase of two points located at the edges of the area obtained by projecting the resolution cell phase from each scatterer within the resolution cell, as shown in Fig. 7. Using this simple model, one can estimate that the null-to-null angular width of the correlation function, $\delta\theta$, is given by

$$\delta\theta \approx \frac{B_{\perp}}{\rho} \approx \frac{\lambda}{\Delta\rho_{\perp}} \tag{58}$$

where B_{\perp} is the projection of the interferometric baseline onto the direction perpendicular to the look direction, and $\Delta \rho_{\perp}$ is the projection of the ground resolution cell along the same direction. This relationship can also be understood in a complementary manner if one considers the interferometric fringes due to two point sources located at the extremes of the projected resolution cell. From elementary optics [42], the nulls in the interference fringe pattern occur when the phase difference, $\phi \approx k B_{\perp} \theta \approx k B_{\perp} \Delta \rho_{\perp}/\rho$, is a multiple of 2π . Rearranging terms, and comparing against (58), one sees that complete decorrelation occurs when the interferometric phase varies by one full cycle across the range resolution cell. In general, due to the Fourier transform relation between illuminated area and correlation distance, the longer the interferometric baseline (or, conversely, the larger the resolution cell size), the lower the correlation between the two interferometric channels.

A more exact calculation results in the following expression for the full interferometric correlation:

$$\gamma = \gamma_N \gamma_G \gamma_Z \tag{59}$$

where γ_G , the geometric correlation, occurs for both surface and volume scattering, and γ_Z , the volume correlation, occurs only for volume scattering. The geometric correlation depends on the system parameters and the observation geometry, and is given by

$$\gamma_G = \frac{\int ds \ d\rho \ W_1(\rho, x) W_2^*(\rho + \delta_\rho, s + \delta_s) \exp\left[i\rho \left(p\kappa_\rho + 2\delta k\right)\right] \exp\left[i\tan \tau_s \kappa_z x\right]}{\int ds \ d\rho \ W_1(\rho, s) W_2^*(\rho, s)}$$
(60)

where δk represents the shift in the wavenumber corresponding to any difference in the center frequencies between the two interferometric channels; δ_{ρ} and δ_{s} are the misregistration between the two interferometric channels in the range (ρ) and azimuth (x) directions, respectively; $W_{i}(\rho, s)$ is the SAR point target response in the range and azimuth directions; and τ_{s} is the surface slope angle in the azimuth direction. In the previous equation, κ_{ρ} and κ_{z} are the interferometric fringe wavenumbers in the range and vertical directions, respectively. They are given by

$$\kappa_{\rho} = \frac{kB_{\perp}}{\rho \tan(\theta - \tau_{c})} \tag{61}$$

$$\kappa_z = \frac{kB_{\perp}\cos\tau_c}{\rho\sin(\theta - \tau_c)} = \kappa_{\rho} \frac{\cos\tau_c}{\cos(\theta - \tau_c)}.$$
 (62)

Equation 60 shows explicitly the Fourier transform relation between the SAR point target response function (the equivalent of the illumination in the vCZ theorem) and the geometric correlation. It follows from this equation that by applying different weightings to the SAR transmit chirp spectrum, thus modifying W(r,x), one can change the shape of the correlation function to reduce phase noise. Figure 20 shows the form of γ_G as a function of the critical baseline, the baseline for which correlation vanishes $(\rho \lambda/\Delta \rho_{\perp})$ from (58), for a variety of spectral weighting windows.

Fig. 20. Baseline decorrelation for various point target response functions.

Notice that if $2\delta k = -\kappa_r$, one obtains $\gamma_G = 1$ [45]. In practice, this can be done by bandpass filtering the signals from both channels so that they have slightly different center frequencies. Unfortunately, this relationship depends on the look angle and surface slope, so that adaptive iterative processing is required in order to implement the approach exactly.

The final contribution to the correlation, γ_Z , is due to volume scattering. The effect of scattering from a volume on the correlation function can be understood based on our previous discussion of the vCZ theorem. From Fig. 21, one sees that the effect of a scattering layer is to increase the size projected range cell, which, according to (58), will result in a decrease of the correlation distance. If the range resolution is given by a delta function, the volume decorrelation effect can be understood as being due to the geometric decorrelation from a plane cutting through the scattering volume perpendicular to the look direction.

Fig. 21. Form of the volumetric decorrelation.

It was shown in [15] that γ_Z can be written as

$$\gamma_Z(\kappa_z) = \int dz \ f(z) \exp\left[-i\kappa_z z\right] \tag{63}$$

provided the scattering volume could be regarded as homogeneous in the range direction over a distance defined by the range resolution. The function f(z), the "effective scatterer probability density function (pdf)", is given by

$$f(z) = \frac{\sigma(z)}{\int dz \ \sigma(z)} \tag{64}$$

where $\sigma(z)$ is the "effective" normalized backscatter cross section per unit height. The term effective is used to indicate that $\sigma(z)$ is the intrinsic cross section of the medium attenuated by all propagation losses through the medium. The specific form for $\sigma(z)$ depends on the scattering medium. Models for this term, and its use in the remote sensing of vegetation height, will be discussed in the applications section of this paper.

IV. INTERFEROMETRIC SAR IMPLEMENTATION

Figure 22 represents an example of an interferometric SAR system for topographic mapping. Several parameters defining the performance and calibration of the interferometer relate directly to radar hardware observables. The relationship of these parameters to the hardware schematic are also important for understanding specific interferometer implementations. For the purpose of discussion, the fan-beam planes of each antenna are assumed to be co-aligned. Other specific hardware implementations have similar elements of radar signal down-conversion and delay. Antennas 1 and 2 represent the apertures of the interferometer. The baseline vector connects the phase centers of the antennas. In relating this figure to the equations above, the following key parameters must be measured:

- Baseline vector, \vec{B} , including length and attitude, for reduction of interferometric phase to height. This parameter translates to knowing the locations of the phase centers of the interferometer antennas typically to the millimeter level.
- Total radar range, ρ , from one of the antennas to the targets, for geolocation. This parameter translates in hardware to knowing the time delays through the composite transmitter and receiver chain typically to a level of a few nanoseconds.
- Differential radar range, $\Delta \rho$, between channels, for image alignment in interferogram formation. This parameter translates to knowing the time delays through the receiver chains, τ_r^1 and τ_r^2 (but not the transmitter chain necessarily) typically to a level of a few nanoseconds.
- Differential phase, $\Delta \phi$, between channels, for determination of the topography. This parameter translates to knowing the phase delays through the receiver chains, ϕ_1 and ϕ_2 to a level of a few degrees or better. It also requires knowing any variations in the phase centers of the antennas, if such exist, for all relevant beam positions and polarizations. It also requires knowing the variations of the phase with incidence angle that constitute the multipath signal, such as scattering of radiated energy off e.g. wings, fuselage, or radome in the aircraft case and booms or other structures on a specific platform.

For a sinusoidal signal injected at Antenna a, the phase measured at the output of receiver a at time t is given by

$$\phi_{i}(t) = \omega_{BB}(t - \tau_{r}^{i}) + \sum_{l=0}^{N-2} \tau_{l}^{i} \left(\sum_{k=l+1}^{N-1} \omega_{k} \right) + \tau_{i} \sum_{l=0}^{N-1} \omega_{l} + \sum_{l=0}^{N-1} \phi_{l}^{i} - \frac{2p\pi}{\lambda} \rho_{i} + \phi_{T}$$
(65)

where ω_{BB} is the baseband frequency given by

$$\omega_0 = \omega_{\text{BB}} + \sum_{l=0}^{N-1} \omega_l \tag{66}$$

 ω_0 is the radio frequency of the sinusoid entering the antenna, τ_i is the delay associated with the antenna amplifier electronics, ω_l is the frequency of the lth mixer in the receiver chain, ϕ_l^i is the random reference phase of the lth mixer of the receiver i chain, τ_l^i is the delay in the lth filter/amplifier of the receiver i chain, N is the number of filter stages of the receiver, λ is the radar wavelength, ρ_i is the range from antenna i to the target, and ϕ_T is the transmitter path induced phase delay. The total time delay through receiver i is

$$\tau_r^i = \tau_i + \sum_{l=0}^{N-1} \tau_l^i \tag{67}$$

By Eq. 65, a signal of general spectral content $S(\omega) \exp(j\omega_0 t)$ ($S(\omega)$ being the baseband signal content) injected at antenna *i*, will suffer a uniform time delay τ_r^i during the down-conversion process, and

constant phase delay ϕ_r^i , to become

$$S(\omega)e^{j\omega_{\rm BB}(t-r_r^i)}e^{j\phi_r^i} \tag{68}$$

where

$$\phi_r^i = \sum_{l=0}^{N-2} \left(\tau_l^i \sum_{k=l+1}^{N-1} \omega_k \right) + \tau_i \sum_{l=0}^{N-1} \omega_l + \sum_{l=0}^{N-1} \phi_l^i$$
 (69)

In this example, the time delay τ_T and phase delay ϕ_T associated with the transmitter chain are common to both receiver channels. Therefore, though the total electronic time delay should be stable or measured to an accuracy that allows adequate geolocation (fraction of a range pixel), the channel time and phase delays of the transmitter cancel in the channel difference, even if they are time-variable.

From the above equations, the interferometric phase difference is

$$\phi = \phi_1 - \phi_2 = -\omega_{BB}(\tau_r^1 - \tau_r^2) + (\phi_r^1 - \phi_r^2)$$
 (70)

$$-\frac{2p\pi}{\lambda}(\rho_1 - \rho_2) \tag{71}$$

If the received baseband signal is adjusted to account for the time delay difference, which can be determined from the data, then the first term in Eq. 71 vanishes, and the interferometric phase difference consists of the desired geometric term and the difference of the receiver phase delays. In general, the receiver phase delays must be measured. This can be done, for example, with the aid of a calibration signal injected in the receive path.

Fig. 22. Definitions of interferometric parameters relating to a possible radar interferometer configuration. In this example, the transmitter path is common to both roundtrip signal paths. Therefore the transmitter phase and time delays cancel in the channel difference.

A. Spaceborne vs. Airborne systems

The following is a comparison of key attributes of spaceborne and airborne interferometric SARs with regard to various applications.

A.1 Coverage

Spaceborne platforms have the advantage of global and rapid coverage and accessibility. While it is possible to build an aircraft system that has moderate (30m) resolution and wide (80 km) swaths, it is not practical to obtain global data rapidly. Because the difference in velocity between airborne systems ($\approx 200 \text{ m/s}$) and spaceborne platforms (> 7000 m/s) is roughly a factor of 30, a spaceborne interferometric map product of comparable swath that takes on the order of a month to derive would take several years in an aircraft. Airspace restrictions can also make aircraft operation extremely difficult in certain parts of the world. In addition, for change mapping, where revisitation of globally distributed sites is crucial to understanding dynamic processes such as ice motion or volcanic deformation, regularly repeating satellite acquisitions are in general more effective.

The role of airborne sensors lies in regional mapping at fine resolution, for a host of applications such as earth sciences, urban planning, and military maneuver planning. The flexibility in scheduling airborne acquisitions, in acquiring data from a variety of orientations, and in configuring a variety of radar modes are key assets of airborne systems that will ensure their usefulness well into the future. The proliferation of airborne interferometers around the world is evidence of this.

A.2 Repeat Observation Flexibility

To construct useful temporal baselines in interferometry, it is desirable to have control over the interval between repeat coverage of a site. An observing scenario may involve monitoring an area monthly, until it becomes necessary to track a rapidly evolving phenomenon such as a landslide or flood. Suddenly, an intensive campaign of observations may be needed twice a day for a period. This kind of flexibility in the repeat period of a platform is quite difficult to obtain with a spaceborne platform. The repeat period must be chosen to accommodate the fastest motion that must be tracked. Thus in the example above, the repeat period must be set to twice per day even though the nominal repeat observation for a spaceborne SAR may be one month. The separation of nadir tracks on the ground is inversely proportional to the repeat period. For a repeat period of n days, the number of orbits executed is $N_O = 86400n/T_p$, where T_p is the orbital period of the platform in seconds. At the equator, the ground track separation is $2\pi R_e/N_O$, where R_e is the radius of the Earth. As the satellite ground tracks become more widely spaced it becomes more and more difficult to target all areas between tracks: the range of incidence angles becomes very large, and range ambiguities and low SNR in the far range begin to limit performance [4]. In any mission design, this trade off between rapid repeat and global accessibility must be made. If global accessibility is of less importance to mapping objectives, then naturally a spaceborne approach becomes more attractive.

Even with a global requirement, for most large scale deformation mapping operations, the fraction of observation time devoted to steady monitoring for deformation processes is far greater than special case monitoring scenarios. It is possible to interrupt nominal operations occasionally with orbital configuration changes that lie within maneuver fuel budgets. The benefits of spaceborne measurements, including measurement accuracy, track repeatability, cost, and coverage often outweigh the potential loss of repeat flexibility.

A.3 Track Repeatability

While aircraft do not suffer as much from temporal observation constraints, most airborne platforms are limited in their ability to repeat their flight track with sufficient control. For a given image resolution and wavelength, the critical baseline for spaceborne platforms is longer than airborne platforms by the ratio of their target ranges, typically a factor in the range of 20-100. For example, a radar operating at C-band at 40 MHz range bandwidth looking at 35 degrees from an airborne altitude of 10 km has a critical baseline of 65 m. Thus, the aircraft must repeat this flight track with a separation distance of fewer than about 30 m to maintain adequate coherence. The same radar at an 800 km spaceborne altitude has a 5 km critical baseline.

The ability to repeat a track depends on both flight track knowledge and track control. GPS technology allows real-time measurement of platform positions at the meter level, but few aircraft can use this accurate information for track control automatically. The only system known to control the flight track directly with inputs from an onboard GPS unit is the Danish EMISAR. Campaigns with this system show track repeatibility of better than 10 m [46].

Despite the typically longer critical baseline from space, spaceborne orbit control is complicated by several factors. Fuel conservation for long duration missions limits the number of trajectory correction maneuvers, allowing the orbit to drift over time. The applied manuever is then difficult to compute correctly because drag and gravitational forces perturb the orbital elements dynamically, making the process of control somewhat iterative, and usually expensive. The ERS satellites for example maintain their orbits to within about 1 km [47].

GPS receivers on spaceborne platforms are allowing kinematic orbit solutions accurate to several tens of meters in real-time [48]. With this knowledge, rapid accurate trajectory corrections will become available, either on the ground or onboard. The TOPEX mission carries a prototype GPS receiver as an experiment in autonomous orbit control [49]. In this experiment, orbital GPS data are sent to the ground for processing, a correction maneuver is computed (then verified by conventional means), and

the correction is uplinked to the satellite. The TOPEX team has been been able to "autonomously" control the orbit to within 1 km, signaling an era of affordably controllable spacecraft for interferometry.

A.4 Motion Compensation

Motion compensation is needed in SAR processing when the platform motion deviates from the prescribed, idealized path assumed (see Fig. 23). The process of motion compensation amounts to a pulse by pulse range-dependent range and phase correction to align pulses over a synthetic aperture as though they were collected on an idealized flight track [4], [38], [50]. If motion compensation is not applied, processed images will be defocused and the image will exhibit distortions of scatterer positions relative to their true planimetric positions. In interferometry, this has two primary consequences: 1) the variation along-track of the scatterer location in range implies that two images forming an interferogram will not, in general, be well-aligned, leading to mis-registration decorrelation and slope-dependent phase errors [38]. 2) defocused imagery implies lower SNR in the interferometric phase measurement. The resulting topographic or displacement map will have a higher level of noise. Since the fuzzy image pixels are correlated, averaging samples to reduce noise will also not be as beneficial.

Accurate airborne interferometric SARs require motion compensation. Over the length of a synthetic aperture, flight path deviations from a linear track can be several meters. This is often the size of a range resolution cell. Platform rotations can be several milliradians. For two-aperture systems, the antennas move together except for rotations of the aircraft, so the image mis-alignment problem is limited to correcting for these rotations. Compensation to a linear flight line is still required to improve focusing. In repeat pass airborne applications, the aperture paths are independent, so misalignment can be quite severe without proper motion compensation. (Velocity differences between repeat paths lead to an analogous along-track compensation correction.)

Fig. 23. Motion compensation geometry illustrated for interferometry. a) Two flight paths and their idealized companions. b) Motion compensation formulation for a dual reference track approach.

A.5 Propagation Effects

The atmosphere and ionosphere introduce propagation phase and group delays to the SAR signal. Airborne InSAR platforms travel below the ionosphere so they are insensitive to ionospheric effects. Spaceborne platforms travel in or above the ionosphere. Both airborne and spaceborne InSARs are effected by the dry and wet components of the troposphere.

Signals from two-aperture InSAR antennas traverse basically the same propagation path, as described previously. The common range delay comprises the bulk of the introduced range error. There is a small differential phase correction arising from aperture separation. Both terms introduce submeter level errors in reconstructed topography (see App. B).

In repeat track systems, the propagation effects can be more severe. The refractive indices of the atmosphere and ionosphere are not homogeneous in space or time. In space, the path delays can be very large, depending on the frequency of the radar (e.g. greater than 50 m for L-band ionospheric path delay). Numerous studies have shown artifacts in repeat-track interferograms that appear to be due to propagation effects [51], [52], [53], [18], [54]. Ionospheric delays are dispersive, so frequency-diverse measurements can potentially help mitigate the effect. Tropospheric delays are non-dispersive, and mimic topographic or displacement effects. There is no means of removing them without supplementary data. Schemes of averaging interferograms to reduce atmospheric noise have been introduced [54], [55], but no systematically valid correction currently exists.

Characteristic	SPACEBORNE	AIRBORNE
Coverage	Global	Regional
Fine Resolution	Costly	Affordable
Repeat Flexibility	Difficult	Natural
Track Repeatability	Difficult/Costly	Difficult/Costly
Motion Compensation	Benign	Necessary
On-board processor	Costly	Affordable
Atmospheric Propagation Effects		
Two-Aperture	Benign	Benign
Repeat Track	Problematic	Problematic
Multi-frequency	No advantage	No advantage
Ionospheric Propagation Effects		N/A
Two-Aperture	Benign	,
Repeat Track	Problematic	
Multi-frequency	Helpful	

A.6 Frequency selection for interferometry

The choice of frequency of an InSAR is usually determined by the electromagnetic phenomena of interest. Electromagnetic energy scatters most strongly from objects matched roughly to the size of the wavelength. Therefore for the varied terrain characteristics on Earth, including leaves high above the soil surface, woody vegetation, very rough lava surfaces, smooth lakes with capillary waves, etc., no single wavelength is able to satisfy all observing desires.

International regulations on frequency allocations also can restrict the choice of frequency. If a particularly wide bandwidth is needed for fine resolution mapping, certain frequency bands may be difficult to use. Other practical matters also determine the frequency, including available transmitter power, allowable antenna size, and cost.

For topographic mapping, where temporal decorrelation is negligible, frequencies can be chosen to image the topography near a desired canopy height. Generally, higher frequencies interact with the leafy crowns and smaller branches strongly, so the inferred interferometric height is near the top of the vegetation canopy. Lower frequencies propagate through the leafy crowns of trees and scatter from larger structures such as branches or ground-trunk junctures, so the inferred height more closely follows the soil surface. This is illustrated in Fig. 24, where the difference in inferred height between C- and L-band TOPSAR data is plotted in image and profile format.

Fig. 24. a) Image of height difference between C- and L-band in Iron Mountain, California. b) Profiles as indicated going from bare fields to forested regions.

For repeat pass interferometry, the frequency selection considerations are complicated. For ice and other relatively smooth surfaces, a shorter wavelength is usually desired because the signal level is generally higher. However, shorter wavelengths tend to interact with vegetation and other small scatterers, which have a greater tendency for movement and changes between observations [18].

B. Airborne Interferometric SAR Systems

Airborne Interferometric SAR systems have been implemented with single-pass across-track and along-track interferometric capabilities as well as for repeat-pass interferometry. The technology for airborne interferometric systems is in most respects identical to the technology applied in standard non-interferometric SAR systems. The needs for accurate frequency and phase tracking and stability of the

channels combined interferometrically are largely identical to the requirements for high-performance image formation in any SAR system combined with the channel tracking required of e.g. polarimetric SAR systems.

Accurate motion and attitude measurements are of key importance in airborne InSAR applications. To avoid significant decorrelation, the two images forming an interferogram must be co-registered with errors that are no more than a small fraction of the resolution cell size. This is generally difficult to achieve in aircraft repeat-pass situations with long flight paths. In the single-pass situation a significant fraction of the motion will be common to both antennas, which reduces motion compensation requirements significantly. Still to determine the location of the individual image points in across-track InSAR systems both the aircraft location and the baseline orientation must be known with great accuracy. Today Global Positioning Systems (GPS) operated in kinematic modes can provide absolute platform locations with decimeter accuracy, and high-performance inertial navigation systems can measure high-frequency motion required for motion compensation. A significant advance in the critical determination of the baseline has been made possible by tightly coupling the INS and GPS. Absolute angle determination with an accuracy of approximately a few thousands of a degree is off-the-shelf technology today (reference to Honeywell 764G needed, how about Litton?).

In addition to the baseline orientation, the baseline length needs to be known. Most single-pass systems developed to date utilize antennas rigidly mounted on the aircraft fuselage. In the recent development of the IFSARE system [56], two antennas were mounted at the ends of an invar frame. Requiring a rigid and stable frame for a two antenna system will, however, severely limit the baseline that can be implemented on an aircraft system. This problem is especially important when a low frequency single-pass interferometer is required. GeoSAR, a system presently being developed by JPL includes a low-frequency interferometer centered at 350 MHz. To achieve a sufficiently long baseline on the Gulfstream G-2 platform, the antennas are mounted in wing tip-tanks. It is expected that, due to the motion of the wings during flight, the baseline is constantly varying. To reduce the collected SAR data to elevation maps the dynamically varying baseline is measured with a laser-based metrology system, which determines the baseline with sub-millimeter accuracy.

For multiple pass airborne systems to be useful it is important that the flight pass geometry can be controlled with precision. Typically, baselines in the range of 10 to 100 m are desired, and it is also important that the baselines are parallel. Standard flight management systems do not support such accuracies. One system which has been specifically modified to support aircraft repeat pass interferometry is the Danish EMISAR system, which is operated on a Royal Danish Air Force Gulfstream G-3. In this system the radar controls the flight-path via the aircrafts Instrument Landing System (ILS) interface. Using P-code GPS as the primary position source, this systems allows a desired flight-path to be flown with an accuracy of typically 10 m or better.

C. Spaceborne Interferometric SAR Experiments

As mentioned in the Introduction, several proof-of-concept demonstration experiments of space-borne interferometric SAR were performed using the repeat-track approach. Li and Goldstein first reported such an experiment using the SEASAT SAR system. While this approach does suffer from the uncertainties due to changes in the surface, and propagation delay effects between the observations and the difficulties of obtaining baseline determination results with precision required for topography mapping, it clearly has the advantage that only one SAR system need to be operating at a time. To demonstrate the capability of this approach on a global scale, the European Space Agency has operated the ERS-1 and ERS-2 satellites in a so-called "tandem mission" approach. The two spacecraft obtained SAR measurements for a significant fraction of the earth's surface with measurements from one spacecraft 1 day after those from the other, with the two spacecraft in a nearly repeat ground track orbital configuration. The 1 day separation in the observations was chosen to minimize the changes mentioned above. A report with examples of the interferometric SAR measurements has been issued

[57]. The detailed quantitative evaluation of this data set has yet to be carried out. However, from some of the preliminary results, one can observe temporal decorrelation in certain regions of the world, especially in heavily vegetated areas, even with the relatively short time separation of I day. In areas where such temporal decorrelation is not significant, it is important to perform an assessment of the quantitative accuracy of the topography data which can be generated with this extensive data set.

To avoid some of the limitations of the repeat-track interferometric SAR experiments, the National Space and Aeronautic Administration, in conjunction with the National Imagery and Mapping Agency, of the US are developing a Shuttle Radar Topography Mission (SRTM). The payload of this mission is based on the SIR-C/X-SAR system which was flown on the shuttle twice in 1994 [58]. This system is currently being augmented by an additional set of C- and X-band antennas which will be deployed by an extendible mast from the shuttle once the system is in orbit. Figure 25 shows the deployed system configuration. The SIR-C/X-SAR radar system inside the shuttle bay and the radar antennas and electronics systems attached to the end of the deployed mast will act as an interferometric SAR system. The length of the mast after deployment, which corresponds approximately to the interferometric SAR baseline, is about 60 meters. The goal of SRTM is to completely map the topography of the global land mass which is accessible from the shuttle orbit configuration (approximately covering 56 degrees South to 60 degrees North) in an 11-day shuttle mission. The C-band system will operate in a ScanSAR mode [59] and will obtain data over an instantaneous swath of about 225 Km. The radar system is based on the SIR-C system with modifications to allow data captured by both interferometric antennas and with simultaneous operation of both a horizontally polarized antenna beam and a vertically polarized antenna beams. By operating the two antenna beams concurrently, it increases the data accuracy and coverage. By combining the data from both ascending and descending orbits, the topography data with a post spacing of about 30 m data expected to have an absolute height measurement accuracy of about 10-15 meters.

Fig. 25. The Shuttle Radar Topography Mission flight system configuration. The SIR-C/X-SAR L, C, and X-band antennas reside in the shuttle's cargo bay. The C and X band radar systems are augmented by receive-only antennas deployed at the end of a 60 m long boom. Interferometric baseline length and attitude measurement devices are mounted on a plate attached to the main L-band antenna structure. During mapping operations, the shuttle is oriented to that the boom is 45 degrees from the horizontal.

A key feature of SRTM is an onboard metrology system to determine the baseline length and orientation between the antenna inside the shuttle bay and the antenna at the tip of the deployed mast. This metrology system is designed to obtain the baseline measurements with accuracies which can meet the absolute topography measurement requirements listed above. As with the two previous flights, the data collected in the mission are stored on onboard tape recorders and upon landing, the more than 100 tapes of SAR data would then be transferred to a data processing system for global topography generation. It is expected that the data processing will take about 1 year to generate the final topography maps. The X-SAR system will also operate in conjunction with the additional X-Band antenna as an interferometric SAR. The instantaneous swath of the X-band system is about 55 Km. While in some areas of the globe, the X-band system will not provide complete coverage, it is expected that the resolution and accuracy of the topography data obtained will be better than those obtained with the C-band system. The results from both systems can be used to enhance the accuracy and/or coverage of the topography results and to study the effects of vegetation on the topography measurements across the two frequencies. At present, this mission is planned to be launched in September, 1999.

The use of spaceborne SAR data for repeat track interferometric surface deformation studies is becoming widespread in the geophysical community [60]. While this approach has uncertainties caused by path delay variability in the atmosphere or ionosphere, it provides the truly unique capability to map small topography changes over large areas. ERS-1 and ERS-2 data are currently routinely used

by researchers to conduct specific regional studies. RADARSAT has been shown to be useful for interferometric studies, particularly using its fine beam mode, but much of the data is limited by relatively large baselines. JERS also has been used to image several important deforming areas. We expect that in the near future, repeat pass interferometry will be possible on a more operational basis using a SAR system dedicated for this purpose.

V. APPLICATIONS

A. Topographic mapping

Radar interferometry is expanding the field of topographic mapping [61], [62], [63], [64], [65], [66], [41], [67], [68], [69]. Airborne optical cameras continue to generate extremely fine resolution (often sub-meter) imagery without the troublesome layover and shadow problems of radar. However, radar interferometers are proving to be a cost-effective method for wide area, rapid mapping applications, and do not require extensive hand-editing and tiepointing. Additionally, these systems can be operated at night in congested air-traffic corridors that are often difficult to image photogrammetrically, and at high altitudes in tropical regions that are often cloud-covered.

A.1 Topographic Strip Mapping

Typical strip-mode imaging radars generate data on arbitrarily long paths in swaths between 6-20 km for airborne systems and 80-100 km for spaceborne systems. These strip digital elevation models can be used without further processing to great advantage. Figure 26 shows a digital elevation model (DEM) of Mount St. Helens imaged by the NASA/JPL TOPSAR C-band interferometer in 1992, years after the eruption that blew away a large part of the mountain (prominently displayed in the figure), and destroyed much of the area. This strip map was generated automatically with an operational InSAR processor. Such rapidly generated topographic data can be used to assess the amount of damage to an area by measuring the change in volume of the mountain from before to after the eruption (assuming a DEM is available before the eruption).

Fig. 26. DEM of Mount Saint Helens generated in 1992 with the TOPSAR C-band interferometer. Area covered is roughly 6 km across track by 20 km along track.

Another example of a strip DEM, generated by the EMISAR system of Denmark is shown in Fig. 27. DEMs such as these are providing the first detailed topographic data base for the polar regions. Because image contrast is low in snow-covered regions, optical stereo mapping can encounter difficulties. A radar interferometer, on the other hand, relies on the same arrangement of the scatterers that comprise the natural imaging surface, and so is quite successful in these regions. However, since radar signals penetrate dry snow and ice readily, the imaged surface does not always lie at the snow-air interface.

Fig. 27. DEM of Askja, Northern volcanic zone, Iceland derived from the C-band EMISAR topographic mapping system. The color variation in the image is derived from L-band EMISAR polarimetry.

Slope estimates such as illustrated in Fig. 28 are useful for hydrological studies and slope hazard analysis. Special care must be taken in computing the slopes from interferometric DEMs because the point-to-point height noise can be comparable to the post spacing. Studies have shown that when this is taken into account, radar-derived DEMs improve classification of areas of landslide-induced seismic risk [69].

Figure 29 illustrates a continental scale topographic strip map. This DEM was generated from the SIR-C L-band system, during the SIR-C/X-SAR mission phase when the Shuttle was operating as a repeat-track interferometer. While the accuracy of repeat-track DEMs is limited by propagation path

Fig. 28. Height (above) and slope (below) maps of Mount Sonoma, CA. This information has been used to assess the risk of earthquake damage induced by landslides.

delay artifacts, this figure illustrates the feasibility of spaceborne global-scale topographic mapping. Figures 30 and 31 illustrate topographic products from ERS and JERS repeat-track interferometry.

Fig. 29. Strip of Topography generated form the SIR-C L-band radar data by repeat track interferometry. The DEM extends from the Oregon/California border through California to Mexico, roughly 1600 km.

Fig. 30. DEM of Mount Etna, Italy generated by ERS repeat track interferometry. Actually 10 images were combined to make this DEM.

Fig. 31. DEM of Mount Unzen, Japan generated by JERS repeat track interferometry.

A.2 Topographic Mosaicks

For many wide-area mapping applications, strip DEMs provide insufficient coverage, so it is often necessary to combine, or "mosaic," strips of data together. In addition to increasing contiguously mapped area, the mosaicking process can enhance the individual strips by filling in gaps due to layover or shadow present in one strip but not in an overlapping strip.

The accuracy of the mosaic and the ease with which it is generated rely on the initial strip accuracies, available ground control, and the mosaicking strategy. Traditional radar mosaicking methods are two-dimensional, assuming no height information. For radar-derived DEMs, a mosaicking scheme that allows for distortions in three dimensions described by an affine transformation, including scale, skew, rotation, and translation, is usually necessary to adjust all data sets to a limited set of ground control points. If the interferometric results are sufficiently accurate to begin with, such as is planned for the Shuttle Radar Topography Mission, mosaicking the data is straightforward.

Figure 32 shows a mosaic of NASA/JPL TOPSAR C-band data acquired over Long Valley, California. The mosaic is posted at 10 m with a spatial resolution of 15 m, and representing the most accurate DEM available of this region. The height accuracy is 3-4 m. Long Valley is volcanically active, and is an area of intense survey and interest. This mosaic is both a reference to track future large scale changes in the shape of the caldera, and a reference with which to generate synthetic fringes for deformation studies.

A.3 Accuracy Assessments

One of the most important aspects of interferometry is in the assessment of DEM errors. Accuracy can be defined in both an absolute and relative sense. The absolute error of a DEM can be defined as the root-mean-square of the difference between the measured DEM and a noise-free reference DEM.

$$\sigma_{\text{h,abs}} = \sqrt{\frac{1}{N_{\text{"DEM"}}}} \sum_{\text{"DEM"}} (z - z_{\text{ref}})^2$$
 (72)

The summation is taken over the DEM extent of interest ($N_{\text{"DEM"}}$ points), so the error can be dependent on the size of the DEM, especially if systematic errors in system parameters are present. For example,

Fig. 32. Long Valley mosaic of TOPSAR C-band interferometric data.

an error in the assume baseline tilt angle can induce a cross track slope error, causing the absolute error to change across the swath.

The relative error can be defined as the standard deviation of the height relative to a noise-free reference DEM.

$$\sigma_{\rm h,rel} = \sqrt{\frac{1}{N_{\rm "box"}}} \sum_{\rm "box"} (z - z_{\rm ref} - (z - z_{\rm ref}))^2$$
 (73)

Note that this definition of the relative error matches the locally scaled interferometric phase noise, given by

 $\sigma_{\mathbf{h}} = \frac{\partial T_h}{\partial \phi} \sigma_{\phi} \tag{74}$

where σ_{ϕ} in the limit of many looks is given by Eq. 56, when the summation box size is sufficiently small. As the area size increases other systematic effects enter into the relative error estimate. Other definitions of relative height error are possible, specifically designed to blend the statistical point-to-point error and systematic error components over larger areas. In this paper, we exclusively use Eq. 73.

Figure 33 illustrates one of the first comparisons of radar data to a reference DEM [19]. The difference between the TOPSAR C-band data and that produced photogrammetrically at finer resolution and accuracy by the Army Topographic Engineering Center (TEC) shows a relative height accuracy of 2.2 m over the TEC DEM. No absolute accuracy assessment was made, and the two DEMs were pre-registered using correlation techniques.

Fig. 33. Difference image between TOPSAR C-band derived DEM and a TEC photogrammetrically-generated reference DEM.

Figure 34 compares a SIRC repeat-pass spaceborne-derived DEM to a TOPSAR mosaic. Errors in this scene are a combination of statistical phase noise-induced height errors, and those due to variability of the tropospheric water vapor through the scene between passes. In fact the major contribution to the 8 m height standard deviation attained for this region (computed using Eq. 73 over the entire scene) was likely to be caused by water vapor contamination. This contrasts with the predicted 2-3 m relative height error obtained from Eq. 74.

Fig. 34. Difference image between SIR-C C-band derived DEM and a TOPSAR mosaic used as a reference DEM.

Figure 35 illustrates an innovation in DEM accuracy assessment using kinematic GPS data. A GPS receiver mounted on a vehicle drove along a radar-identifiable road within the DEM. The trace of the GPS points was cross-correlated with the TOPSAR image to register the Kinematic data to the DEM. Measured and predicted relative height errors are shown in the figure [70]. A similar approach is planned for assessing the absolute errors of the SRTM global DEM, using kinematic surveys of several thousand km around the world.

Fig. 35. Illustration of the use of kinematic GPS surveys in determining the absolute and relative error of a radar-derived DEM. Curve shows the standard deviation of the radar height relative to the GPS, and its predicted value. Statistical height error estimates derived from the correlation track the measured local statistical height errors extremely well.

B. Crustal Dynamics Applications

Differential Interferometry has generated excitement in the Earth science community in applications to the study of fault mechanics, long period seismology, and volcanic processes. The surface displacements measured by differential interferometry are uniquely synoptic, capturing subtle features of change on the Earth that are distributed over wide areas. There is a growing literature on the subject, which recently received a comprehensive review [60].

The first publication in the open literature demonstrating the method showed centimetric swelling of irrigated fields in Imperial Valley, California [14]. The data were acquired by the L-band SAR aboard SEASAT, and illustrated the power of the method. It did not receive much attention, however, until the ERS C-band SAR captured the displacement field of the Landers M=7.2 earthquake of 1992. The broad and intricate fringe patterns of this large earthquake representing the net motion of the Earth's surface from before to after the earthquake graced the cover of Nature [71] and set the stage for rapid expansion of applications of the method. Since the Nature article, differential interferometry has been applied to co-seismic [72], [73], [74], [76], [75], [77], post-seismic [79], [78] and aseismic tectonic events [80], volcanic deflation [81], [18], [82], ground subsidence and uplift from oil and water pumping [83], [84], and landslide tracking [85]. The most important contributions by differential interferometry lie in areas where conventional geodetic measurements are limited. Associated with surface deformation, the correlation measurements have been used to characterize zones where surface disruption was too great for interferometry to produce a meaningful displacement estimate [86]. In addition to demonstration of science possibilities, the relatively large volume of data acquired by ERS-1, ERS-2, JERS-1, SIR-C, and RADARSAT has allowed for a fairly complete assessment of interferometric potential for these applications.

Co-seismic displacements in the far field away from the faults are generally well understood mechanically by seismologists. The far-field signature of the Landers co-seismic displacements mapped by ERS matched well with a model calculation based on elastic deformation of multiple faceted plate dislocations embedded in an infinite half space [71]. The GPS network at Landers was dense enough to capture this far field pattern, so in that sense the radar measurements were not essential to understanding the coseismic signature of the earthquake. However, the radar data showed more than simply the co-seismic displacement field. What appears to be severe cracking of the surface into tilted facets was reported by Peltzer et al. [74] and Zebker et al. [72]. Tilted features are areas whose surface properties remain intact spanning a deformation event. Thus their fringe pattern changes relative to their surroundings, but they remain correlated. Peltzer explained the tilted patches near the main Landers ruptures as due to shear rotation of the sideward slipping plates, or grinding of the surface at the plate interface. The cracked area described by Zebker et al., farther from the rupture zone, remains unexplained.

The M=6.3 Eureka Valley, CA Earthquake in 1993 is an example of an application of interferometry to a locally uninstrumented site where important science insight can be derived. Two groups have studied this earthquake, each taking a different approach. Peltzer and Rosen [75] chose to utilize all available data to construct a geophysically consistent model that explained all the observations. Those observations included the differential interferogram, the seismic record, which included an estimate of fault plane orientation and depth of the slip, field observations, and geologic context provided by fault maps of eastern California. The seismic record predicted a fault plane orientation relative to North, known as "strike", that was aligned with faulting history for normal faults (i.e. faults whose motion is principally due to separation of two crustal regions) in the area. Without further data, no further insight into the fault mechanism would be possible. However, Fig. 36 shows that the NNW orientation of the subsidence ellipse measured in the interferogram is not consistent with the simple strike mechanism oriented NNE according to the seismic record. Peltzer resolved the conflict by allowing for a spatially variable distribution of slip on the fault plane, originating at depth to the north and rising on the fault plane to break the surface in the south. Fresh, small surface breaks in

the south were observed in the field.

Fig. 36. Subsidence caused to an M=6.3 earthquake along a normal fault in Eureka Valley, California imaged interferometrically by ERS-1. The interferometric signature combined with the seismic record allowed unique interpretation of variable slip along the fault.

Massonnet and Feigl [52] chose to invert the Eureka Valley radar measurements unconstrained by the seismic record, and with a simple uniformly slipping fault model. The inferred model did indeed match the observations well, but predicted a depth, extent and orientation of slip that differed from the seismic record.

Post-seismic activity is measured conventionally by seismicity and displacement fields inferred from sparse geodetic measurements. The post-seismic signature at Landers was studied by two groups using interferometry [79], [78]. Peltzer et al. [79] formed differential interferograms over a broad area at Landers, capturing the continued slip of the fault in the same characteristic pattern as the co-seismic signal, as well as localized but strong deformation patterns where the Landers fault system was disjoint. Peltzer interpreted these signals, which decreased in a predictable way with time from the co-seismic event, as due to pore fluid transfer in regions that had either been compressed or rarefied by the sheer motion on disjoint faults. Material compressed in the earthquake has a fluid surfeit compared to its surrounding immediately after the event, so fluids tend to diffuse outward from the compressed region in the post-seismic period. Conversely, at pull-apart regions, a fluid deficit is compensated post-seismically by transfer into the region. Thus the compressed region deflates, and the pull-apart inflates, as observed.

GPS measurements of post-seismic activity at Landers were too sparse to detect these local signals, and seismometers cannot measure slow deformation of this nature. This is prime example of geophysical insight into the nature of lubrication at strike-slip faults that eluded conventional measurement methods.

A-seismic displacements, that is slip along faults that does not generate detectable seismic waves, have been measured on numerous occasions in the Landers area, and elsewhere in the Southern California San Andreas sheer zone. Sharp displacement discontinuities in interferograms indicate shallow creep signatures along a fault (Fig. 37). Creeping faults may be relieving stress in the region, and understanding their time evolution is important to understanding seismic risk.

Another location where aseismic slip has been measured is along the San Andreas fault. At Parkfield, California, a segment of the San Andreas Fault is slipping all the way to the surface, moving at the rate at which the North American and Pacific tectonic plates themselves move. To the north and south of the slipping zone, the fault is locked. The transition zone between locked and free segments is just northwest of Parkfield, and the accumulating strain, coupled with nearly regular earthquakes spanning over 100 years, has led many to believe that an earthquake is imminent. Understanding the slip distribution at Parkfield, particularly in the transition zone where the surface deformation will exhibit variable properties, can lead to better models of the locking / slipping mechanisms. New work with ERS data, shown in Fig. 37, has demonstrated the existence of slip [80] but the data are not sufficiently constrained to model the mechanisms.

Fig. 37. Aseismic slip along the San Andreas Fault near Parkfield, California, imaged interferometrically by ERS-1.

Inter-seismic displacement fields have never been measured to have local signatures near faults. The sensitivity of the required measurement, and the variety of spatial scales that need to be examined are ideally suited to a properly designed InSAR system. The expectation is that interferometry will provide the measurements over time and space that are required to map interseismic strain accumulation associated with earthquakes.

Active Volcanic processes have been measured through deflation measurements and through decorrelation of disrupted surfaces. While Massonnet et al. [81] showed up to 12 cm of deflation at Mount Etna over a 3 year period, Rosen et al. [18] demonstrated over 10 cm in 6 months at an active lava vent on Kilauea volcano in Hawaii. Zebker et al. [86] showed that lava breakouts away from the vent itself decorrelated the surface, and from the size of the decorrelated area, an estimate of the lava volume could be obtained.

Decorrelation processes may also be useful as disaster diagnostics. Figure 38 shows the signature of decorrelation due to the Kobe earthquake as measured by the JERS-1 radar. Field analysis of the decorrelated regions shows that areas where buildings were located on landfill collapsed, whereas other areas that did not decorrelate were stable. Vegetation is also partially decorrelated in this image, and an operational monitoring system would need to distinguish expected temporal decorrelation, as in trees, from disaster related events.

Fig. 38. Decorrelation in the destroyed areas of Kobe city due to the 1995 M=6.8 earthquake. Areas where structures were firmly connected to bedrock remained correlated, while structures on sandy areas of liquefaction were destroyed and decorrelated in the imagery.

C. Glaciers

The ice sheets of Greenland and Antarctica play an important role in the earth's climatic balance. Of particular importance is the possibility of a significant rise in sea level brought on by a change in the mass balance of, or collapse of, a large ice sheet [88]. An understanding of the processes that could lead to such change is hindered by the inability to measure even the current state of the ice sheets.

Topographic data are useful for mapping and detecting changes in the boundaries of the individual drainage basins that make up an ice sheet [89]. Short-scale (i.e., a few ice thicknesses) undulations in the topography are caused by obstructions to flow created by the basal topography [89], [90]. Therefore, surface topography can be used to help infer conditions at the bed [91] and high resolution DEMs are important for modeling glacier dynamics. Although radar altimeters have been used to measure absolute elevations for ice sheets, they do not have sufficient resolution to measure short-scale topography. As a result, there is little detailed topographic data for the majority of the Greenland and Antarctic ice sheets.

Fig. 39. Ice velocity map draped on topography of Storstrømmen Glacier in Greenland. Both velocity and topography were generated by ERS interferometry. Ice velocity vectors show that the outlet of the glacier is blocked from flow. In addition to aiding visualization of the ice flow, topographic maps such as this are an important measurement constraint on the mass balance, as changes in topographic height relate to the flow rate of ice from the glacier to the sea.

Ice-flow velocity controls the rate at which ice is transported from regions of accumulation to regions of ablation. Thus, knowledge of the velocity and strain rate (i.e., velocity gradient) are important in assessing mass balance and in understanding the flow dynamics of ice sheets. Ground-based measurements of ice-sheet velocities are scarce because of logistical difficulties in collecting such data. Ice-flow velocity has been measured from the displacement of features observed in sequential pairs of visible [92], [93] or SAR images [94], but these methods do not work well for the large, featureless areas that comprise much of the ice sheets. Interferometric SAR provides a means to measure both detailed topography and flow velocity.

C.1 Ice Topography Measurement

The topography of ice sheets is characterized by minor undulations with small surface slopes, which is well suited to interferometric measurement. While the absolute accuracy of intererferometric ice-sheet

Fig. 40. Horizontal velocity field plotted over the SAR amplitude image of the Ryder Glacier. Contour interval is 20 m/yr (cyan) for velocity less than 200 m/yr and is 100 m/yr (blue) for values greater than 200 m/yr. Red arrows indicate flow direction and have length proportional to speed.

topography measurements is generally poorer than that of radar (for flat areas) or laser altimeters, an interferometer is capabable of sampling the ice sheet surface in much greater detail. While not useful for direct evaluation of ice sheet thickening or thinning, such densely sampled DEMs are useful for studying many aspects of ice sheet dynamics and mass balance.

The Canadian Center for Remote Sensing has used its airborne SAR to map glacier topography on Bylot Island in the Canadian Arctic [95]. The NASA/JPL TOPSAR interferometer was deployed over Greenland in the May 1995 to measure ice-sheet topography.

Repeat-pass estimation of ice-sheet topography is slightly more difficult as the motion and topographic fringes must first be separated. Kwok and Fahnestock [68] demonstrated that this separation can be accomplished as a special case of the three-pass approach. For most areas on an ice sheet, ice flow is steady enough so that it yields effectively the same set of motion-induced fringes in two interferograms with equal temporal baselines. As a result, two such interferograms can be differenced to cancel motion, yielding a topography-only interferogram that can be used to create a DEM of the ice-sheet surface. Joughin et al. [37], [96] applied this technique to an area in western Greenland and obtained relative agreement with airborne laser altimeter data of 2.6 m.

With topography isolated by double differencing, the motion-topography separation can be completed with an additional differencing using the topography-only interferogram and either of the original interferograms (see 3-pass section) to obtain a motion-only interferogram.

C.2 Ice Velocity Measurement

Goldstein et al. [12] were the first to apply repeat-pass interferometry to the measurement of ice motion when they used a pair of ERS-1 images to map ice flow on the Rutford Ice Stream, Antarctica. With the availability of ERS data, the ability to interferometrically measure ice-sheet motion is maturing rapidly as indicated by a number of recent publications [1-25]. Joughin and et al. [97] and Rignot et al. [98] studied ice-sheet interferograms created from long strips of imagery from the west coast of Greenland sheet that exhibited complex phase patterns due to ice motion. Hartl et al. [105] observed tidal variations in interferograms of the Hemmen Ice Rise on the Filchner-Ronne Ice Shelf. Kwok and Fahnestock [68] measured relative motion on an ice stream in northeast Greenland. The topography and dynamics of the Austofonna Ice Cap, Svalbards has been studied using interferometry by Unwin and Wingham [110].

Without accurate baseline estimates and knowledge of the constant associated with phase unwrapping, velocity estimates are only relative and are subject to tilt errors. To make absolute velocity estimates and improve accuracy, ground-control points are needed to accurately determine the baseline and unknown phase constant. In Greenland the ice sheet is surrounded by mountains so that is often possible to estimate the baseline using ground-control points from stationary ice-free areas. When the baseline is fairly short (i.e., < 50 m), baseline estimates are relatively insensitive to the ground-control height error, allowing accurate velocity estimates even with somewhat poor ground-control [99]. For regions deep in the interior of Greenland and for most of Antarctica, which has a much smaller proportion of ice-free area, ground-control points often must be located on the ice sheet where the velocity of the points must also be known. While such in situ measurements are difficult to make, four such points yield a velocity map covering tens of thousands of square kilometers.

Interferograms acquired along a single-track are sensitive only to the radar line-of-sight component of the ice-flow velocity vector. If the vertical component is ignored or at least partially compensated for using surface-slope information [99], then one component of the horizontal velocity vector can be

measured. If flow direction can be determined the full flow vector can be determined. Over limited areas flow direction can be inferred from features visible in the SAR imagery such as shear margins. Flow direction can also be estimated from the direction of maximum averaged (i.e., over scales of a several kilometers) downhill slope [89]. Either of these estimates of flow direction have poor spatial resolution. Even when the flow direction is well known, accuracy of the resulting velocity estimate is poor when the flow direction is close to that of the along-track direction where there is no sensitivity to displacement. As a result, the ability to determine the full three-component flow vector from data acquired along a single satellite track is limited.

In principle, direct measurement of the full three-component velocity vector requires data collected along three different satellite track headings. These observations could be acquired with a SAR that can image from either side (i.e., a north/south looking instrument). Current spaceborne SARs, however, acquire interferometric data typically from a north-looking configuration, with the exception of a short duration south-looking phase for RADARSAT. Except for this brief episode, it is not possible to obtain north- and south-looking coverage at high latitudes (above 80 degrees) so that direct comprehensive measurement is not possible over large parts of Antarctica.

With the assumption that ice flow is parallel to the ice-sheet surface, it is possible to determine the full three-component velocity vector using data acquired from only two directions and knowledge of the surface topography. Such acquistions are easily obtained using descending and ascending satellite passes. This technique has been applied by Joughin et al. to the Ryder Glacier Greenland [100] (see Fig. 40). Mohr et al. [104] have also applied the surface-parallel flow assumption to derive a detailed three-component velocity map of Storstrommen Glacier in northeastern Greenland.

With the surface-parallel flow assumption, small deviations from surface-parallel flow (i.e., the submergence and emergence velocity) are ignored without consequence for many glaciological studies. These variations from surface parallel flow, however, do contain information on local thickening and thinning rates. Thus, for some ice sheet studies it is important to collect data from three directions where feasible.

C.3 Glaciological Applications

As measurement techniques mature, interferometry is transitioning from a stage of technique development to one where it is a method routinely applied for ice-sheet research. One useful interometry application is in monitoring outlet glacier discharge. A substantial portion of the mass loss of the Greenland and Antarctic ice sheets results from discharge of ice through outlet glaciers. Rignot [106] used estimates ice thickness at the grounding line and interferometric velocity estimates to determine discharge for several glaciers in northern Greenland. Joughin et al. [108] have measured discharge on the Humboldt and Petermann Glaciers in Greenland by combining interferometrically measured velocity data with ice thicknesses measured with the University of Kansas airborne radar depth sounder.

Because the ice sheets have low surface slopes, grounding line positions, the boundaries where an ice sheet meets the ocean and begins to float, are highly sensitive to thickness change. Thus, changes in grounding line position should provide early indicators of any thickening or thinning caused by global or local climate shifts. Goldstein et al. mapped the location of the grounding line of the Rutford Ice Stream using a single interferometric pair. Rignot [102] developed a three-pass approach that improves location accuracy to a few tens of meters. He has applied this technique to locate grounding lines for several outlet glaciers in Northern Greenland.

Fig. 41. Grounding line time series (Courtesy: E. Rignot; Copyright Science), illustrating the retreat of Pine Island Glacier.

Little is known about the variability of flow speed of large outlet glaciers and ice streams. Using ERS-1 tandem data, Joughin et al. [101] observed a mini-surge on the Ryder Glacier, Greenland.

They determined that speed on parts of the glacier of glacier increased by a factor of three or more and then returned to normal over a period of less than seven weeks. Mohr et al. [104] have observed a dramatic decrease in velocity on Storstrommen glacier, Greenland after a surge.

C.4 Temperate Glaciers

Repeat-pass interferometric measurements of temperate glaciers can be far more challenging than those of ice sheets. Temperate glaciers are typically much smaller and steeper, making them more difficult to work with interferometrically. Furthermore, many temperate glaciers are influenced strongly by maritime climates resulting in high accumulation rates, frequent storms, and higher temperatures that make it difficult to obtain good correlation. Nevertheless measurements have been made on temperate glaciers. Rignot used repeat-pass SIR-C interferometry to study topography and ice motion on the San Rafael Glacier, Chile [103]. Mattar et al. [109] obtained good agreement between their interferometric velocity estimates and in situ measurements.

D. Ocean Mapping

The along-track interferometric (ATI) SAR approach can be used to measure motion of targets within the SAR imagery. The first application of this technique was a proof-of-concept experiment in the mapping of tidal ocean surface current using an airborne ATI SAR ([11]. In that experiment, interferometric SAR signals were obtained from two antennas which were attached near the fore and the aft portions of the NASA DC-8 aircraft fuselage. While one of the antennas was used for radar signal transmission, both of the antennas were used for echo reception. Interferometric measurements were obtained by combining the signals from the fore and the aft antennas by 'shifting', in the alongtrack dimension, the signals from the two antennas such that the signals were overlayed when the two antennas were at approximately the same along-track path location. For the DC-8 aircraft flight speed and the spatial separation of the fore and aft antennas, the aft antenna data were obtained about 0.1 sec after the fore antenna. The interferoemtric phase signals measured correspond to the movement in the ocean surface between the 0.1 sec interval. Adjustments in the data processing were also made to remove effects due to random aircraft motion and aircraft attitude deviations from a chosen reference. The interferometric phase signals were then averaged over large areas of the San Francisco Bay. The resulting average phase measurements were shown to correspond well to those expected due too tidal motion in the ocean surface during the experiment. The tidal motion detected was about 1 m/s, which was consistent with the in situ tidal data available and the ATI SAR measurement accuracy, after the large area averaging, was in the range of 10 cm/s. Figure 42 shows results from a similar ATI experiment conducted at Mission Bay, San Diego, California [113]. The flight tracks were oriented in several directions, to measure different components for the velocity field (the ATI instrument measures only the radial component of motion). In particular note that in Fig. 42a, the wave patterns are clearly visible because the waves are propagating away from the radar toward the shore. In Fig. 42b on the other hand, the waves are propagating orthogonal to the radar look direction, so only the turbelent breaking waves contribute to the radial velocity.

Fig. 42. Example of ocean currents measured by along-track SAR interferometry. Flight direction of the radar is from left to right in each image, so panels a-d show different look aspects of the wave patterns propagating to shore.

Goldstein et al. [11] applied this technique to derive direct, calibrated measurements of ocean wind wave directional spectrum. This proof-of-concept experiment was performed in conjunction with the Surface Wave Process Program experiment. Instead of averaging the phase measurements over large areas, the phase measurements obtained for the intrinsic SAR resolution elements were used to measure the displacement of the ocean surface. Typically, this displacement is the algebraic sum of the small displacements of the Bragg waves, such as the phase velocity of the Bragg waves themselves, the

orbital velocity associated with the swell upon which they ride, and any underlying surface current that may be present. In this experiment, the orbiting motion components due to the wind waves are separated from the Bragg phase velocity and the ocean currents on the basis of the spatial frequencies. The Bragg and the ocean current velocities are usually steady over large areas, whereas the swell is composed of higher spatial frequencies that of interest in the ocean wave spectra measurements. It should be noted that the Bragg waves are not imaged directly as waves, rather they are the scatterers providing the radar return. Because the intereferometer measures directly the line-of-sight velocity, independent of such variables as radar power, antenna gain, surface reflectivity, etc. it enables the determination of the actual height of the ocean waves via linear wave theory. Goldstein et al compared the ocean wave spectra results from the interferometric SAR approach to other conventional in situ measurements and obtained reasonable agreements. Unfortunately, the data set reported was limited to one oceanic condition and more extensive data sets are required to ascertain the effectiveness of this remote sensing technique for ocean wave spectra measurements. Others have studied the utility of ATI measurements [111], [112].

E. Vegetation Algorithms

As discussed above, vegetation canopies have two effects on interferometric signals: first, the mean height reported will lie somewhere between the top of the canopy and the ground; second, the interferometric correlation coefficient will decrease due to the presence of volume scattering.

The first effect is of great importance to the use of InSAR data for topographic mapping since, for many applications, the bare-earth heights are desired. It is expected that the reported height depends on the penetration characteristics into the canopy, which, in turn, depends on the canopy type, the radar frequency, and the incidence angle.

The first reported values of the effective tree height for interferometry was made by Askne et al. [114] [115], using ERS-1 C-band repeat pass interferometry over boreal forests in northern Sweden. For very dense pine forests, whose average height was approximately 16 m, the authors observed effective tree heights varying between 3.4 m and 7.4 m. For mixed Norway Spruce (average height 13 m) and Pine/Birch (average height 10 m) forests, the authors observed effective heights varying between 0 m and 6 m. The bulk of the measurements were not very dependent on the interferometric baseline, although the lowest measurements were obtained for the case with the lowest correlation, indicating that the effect of temporal decorrelation could have affected the reported height: the reported height will be due to the scatterers which do not change between passes, such as trunks and large branches or ground return.

To separate the effect due to penetration into the canopy and temporal decorrelation, it is necessary to examine data collected using single-pass multi-aperture interferometry. Rodríguez et al. [116] collected simultaneous InSAR and laser altimeter data over mixed coniferous forests in southern Washington State, using the JPL TOPSAR interferometer and the NASA GSFC laser profilometer, respectively. Figure 43 shows the laser determined canopy top and bottom together with the InSAR estimated height over a region containing mature stands as well as clear-cuts exhibiting various stages of regrowth. As can be seen from this figure, even for mature forest stands, the InSAR height is approximately half-way between the canopy top and the ground, consistent with the results obtained by Askne et al. This indicates that the observed effects are largely due to penetration into the canopy, and not due to temporal decorrelation. Rather, Rodríguez et al. propose that the bulk of the penetration occurs through gaps in the canopy, a result which is consistent with the decorrelation signature presented below. The results of both Askne et al. and Rodríguez et al. show that penetration into boreal or mixed coniferous forests is significantly higher than that expected using laboratory/field measurements of attenuation from individual tree components, leading to the conclusion that the canopy gap structure (or the area fill factor) plays a leading role in determining the degree of penetration.

The effect of volumetric scattering on the correlation coefficient was also examined by Askne et al.,

Fig. 43. Profiles of canopy extent as measured by the Goddard Space Flight Center Airborne Laser Altimeter.

and a simple electromagnetic model assuming a homogeneous cloud scatterer model and an area fill factor was presented. Using this model, attenuation and area fill parameters could be adjusted to make the model agree with the effective tree height. However, the predicted decorrelation could not be compared against measurements due to the contribution of temporal decorrelation.

Treuhaft et al. [118] used a similar parametric single layer homogeneous canopy model (not including area fill factors) to invert for tree height and ground elevation using same-pass interferometric data over a boreal forest in Alaska. The number of model parameters was greater than the number of available observations, so assumptions had to be made about the medium dielectric constant. While measurements were made during thaw conditions, it was observed that better agreement with ground truth was obtained if the frozen dielectric constant (resulting in smaller attenuation) was used in the model. The results for the inversion of tree height and elevation are shown in Fig. 44. In general, good agreement is observed if the frozen conditions dielectric constant is used, but the heights are overestimated if the thawed dielectric constant is used. This difference may indicate the need for an area fill factor or canopy gap structure, as advocated by Askne et al. and Rodríguez et al., or the inclusion into the model of ground trunk interactions (Treuhaft, private communication, 1997), which would lower the canopy phase center.

Fig. 44. Inversion of tree height and elevation. Courtesy R. Treuhaft. Copyright Radio Science.

In an attempt to overcome what are potentially oversimplifying assumptions about the vegetation canopy, Rodríguez et al. [116] introduced a non-parametric method of estimating the effective scatterer standard deviation using the volumetric decorrelation measurement. They showed that the effective scatterer variance (i.e., the normalized standard deviation of the radar backscatter, including variations due to intrinsic brightness and attenuation, as a function of height), σ_Z^2 , could be estimated from the volumetric correlation γ_Z by means of the simple formula

$$\sigma_Z^2 = \frac{2(1 - \gamma_Z)}{\kappa_Z^2} \tag{75}$$

Rodríguez et al. hypothesized that if, at high frequencies, the dominant scattering mechanism into the canopy was geometric (i.e., canopy gaps), this quantity should be very similar to the equivalent quantity derived for optical scattering measurements, since in both cases the cross section is proportional to the geometric cross section, and the gap penetration is frequency independent. In fact, Figure 43 shows that this is observed for the laser and InSAR data collected over Washington State. Rodríguez et al. speculated that a simple scaling of the estimated scatterer standard deviation might provide a robust estimate of tree height. That this is in fact the case is shown in Figure 45, where measured tree heights are compared against estimated tree heights.

Fig. 45. Estimated scatterer standard deviation scaled empirically, compared to tree height derived by laser altimeter. Agreement is very good.

Summarizing, it is clear that significant penetration into forest canopies is observed in InSAR data, and it is speculated that the dominant mechanism is due to penetration through gaps in the canopy, although other mechanisms, such as ground-trunk interactions, may also play a significant role. Current research (Hensley, Rodríguez and Treuhaft, personal communications, 1998) focuses in the characteri-

zation of penetration characteristics over other vegetation types, the study of the frequency dependence of penetration, and the improvement of inversion techniques for canopy parameter estimation.

F. Terrain Classification Using InSAR Data

The use of interferometric data for terrain classification is relatively new. Two basic approaches have been used for terrain classification using InSAR: 1) classification using multi-temporal repeat pass interferometric data; and, 2) classification using simultaneous collection of both InSAR channels. The idea of using multi-temporal repeat pass data is to make use of the fact, first documented by Zebker and Villasenor [44], that different types of terrains have different temporal correlation properties due to a varying degree of change of the scatterer characteristics (position and electrical) between data takes. Zebker and Villasenor found, using SEASAT data over Oregon and California, that vegetated terrain, in particular, exhibited an interferometric correlation which decreased almost linearly with the temporal separation between the interferometric passes. These authors, however, did not use this result to perform a formal terrain classification.

A more systematic study of the temporal correlation properties of forests was presented by Wegmuller and Werner [119], using ERS-1 repeat pass data. By examining a variety of sites, they found that urban areas, agriculture, bushes, and forest had different correlation characteristics, with urban areas showing the most coherence between passes and forests the least (water showed no correlation between passes). When joint correlation and brightness results are plotted for each class (see Fig. 46), the different classes tend to cluster, although some variation between data at different times is observed.

Fig. 46. Classification space showing image brightness vs. interferometric correlation. Terrain types cluster as indicated.

Based on their 1995 work, Wegmuller and Werner [120] presented a formal classification scheme based on classification based on the interferometric correlation, the backscatter intensity, the backscatter intensity change, and a texture parameter. A simple classifier based on setting characteristic independent intervals for each of the classification features was used. The typical class threshold settings were determined empirically using ground truth data. Classification results for a test site containing the city of Bern, Switzerland, were presented (see Figure 47) and accuracies on the order of 90% were observed for the class confusion matrix.

Fig. 47. Classification of Bern, Switzerland using L- and C- band SIR-C interferometric time series data to distinguish features.

The use of same-pass InSAR data for classification was presented in Rodríguez et al. [117], using the C-band JPL TOPSAR instrument over a variety of sites. Unlike multi-temporal data, same-pass InSAR data does not show temporal decorrelation and the feature vectors used for classification must be different. To differentiate between forested and non-forested terrain, these authors estimated the volumetric decorrelation coefficient, γ_Z , presented above to estimate scatterer standard deviations to be used as a classification feature. In addition, the radar backscatter, the rms surface slope, and the brightness texture were used in a Bayesian classification scheme which used mixtures of Gaussians to characterize the feature vector multi-dimensional distributions. Four basic classes (water, fields, forests, and urban) were used for the classification, and an evaluation based on multiple tests sites in California and Oregon was presented. An example of the results for the San Francisco area are shown in Figure 48.

Rodríguez et al. found that classification accuracies in the 90% level were generally obtained, although significant ambiguities could be observed under certain conditions. Specifically, two problems were observed in the proposed classifications scheme: 1) sensitivity to absolute calibration errors between sites; and, 2) ambiguities due to changes in backscatter characteristics as a function of incidence

Fig. 48. Classification of San Francisco using JPL TOPSAR image brightness, inteferometric correlation, and topographic height and slope.

angle. The effects of the first problem were apparent in the fact that same-site classification always yielded much higher classification accuracies than classification collected for similar sites at different times, probably due to changes in the instrument absolute calibration. The second problem is more fundamental: for small incidence angles (up to about 25°) water can be just as bright as fields, exhibits similar texture and no penetration, causing systematic confusion between the two classes. However, if the angular range is restricted to be greater than 30°, this ambiguity is significantly reduced due to the rapid drop-off of the water backscatter cross section with incidence angle.

Based on these early results, we conclude that InSAR data, although quite different in nature from optical and polarimetric SAR data, shows potential to be used for terrain classification using both multi-temporal and same-pass data. More work is needed, however, to fully assess the ultimate potential of this technique both in accuracy and in determining the classes that can ultimately be separated.

VI. OUTLOOK

Over the past 2 decades, there has been a continuing maturing of the technology for interferometric SAR systems with an associated impressive expansion of the potential applications of this remote sensing technique. One major area of advance is the overall understanding of the system design issues and the contribution of the various sources of uncertainties to the final geophysical parameter measured by an interferometric SAR. These improvements allow systematic approaches to the design, simulation and verification of the performance of interferometric SAR systems. We witnessed the changes from analog signal processing techniques to automated digital approaches, which significantly facilitated the utility of the data products as well as improved on the accuracy and repeatability of the results. Several airborne interferometric SAR systems are currently routinely deployed to provide high resolution topography measurements as well as other data products for geophysical studies. Finally, the spectrum of applications of the interferometric SAR data to multiple scientific disciplines has continued to broaden with an expanding publication of the results from proof-of-concept experiments across these disciplines.

With these advances, the use of spaceborne interferometric SAR systems will be the "approach of choice" for high resolution topography mapping on a regional as well as a global scale. The continuing improvements in the technologies for spaceborne radar systems and the associated data processors will make such an approach more affordable and efficient. We speculate that in the next decade there will be additional spaceborne missions which will provide higher resolution and better height accuracy topography data than those expected for the SRTM mission. Obviously, the key issue of the influence of surface cover, such as vegetation, on the topography results from SARs should be pursued further to allow a better understanding of the relation of the results to the topography of the bare earth.

Airborne interferometric SARs are expected to play an increasing role supplying topographic data in the form of digital elevation models to a variety of users requiring regional scale topographic measurements. The relatively quick processing of InSAR data compared to optical stereo processing makes InSAR attractive from both schedule and cost considerations. We expect this to become the method of choice for a variety mapping applications typically handled by photogrammetric techniques. Newer systems are expected to increase the accuracy and utility of airborne InSAR systems by increasing the bandwidth to achieve higher resolution, moving to lower frequencies, as with the GeoSAR system being developed JPL for sub-canopy mapping, and to systems which are both fully polarimetric and interferometric to exploit the differential scattering mechanisms exhibited by different polarizations [141].

We also speculate that the use of repeat track observations of interferometric SAR for minute surface deformation will become an operational tool for researchers as well as other civilian users to study geophysical phenomena associated with earthquakes, volcanoes, etc. We expect that the results from long term studies using this tool will lead to a significantly better understanding of these phenomena. This improvement will have a strong impact on earth science modeling and the forecasting of natural hazards. As described in Section IV-A.5, the changes in the atmosphere (and the ionosphere) will continue to affect the interpretation of the results. However, by combining data from long time series, it is expected that these effects will be minimized. In fact, we speculate that, once these effects can be isolated from long duration observations, the changes in the atmospheric and ionosphere conditions can become geophysical observations themselves. These subtle changes can be measured with spatial resolutions currently unavailable from on-going spaceborne sensors, and they, in turn, can be valuable input to atmospheric and ionospheric studies.

Future SAR missions optimized for repeat-pass interferometry should allow mapping of surface topography and velocity over entire Greenland and Antarctic ice sheets providing data vital to improving our understanding of dynamics that could lead to ice-sheet instabilities and to determining the current mass balance of the ice sheets. We expect these applications to become routine for glaciology studies.

While large scale application of InSAR data to the areas described above has been hampered by the lack of availability of optimized interferometric data to the science community, we expect this situation to improve significantly in the upcoming decade with the advent of spaceborne SAR systems with inherently phase stable designs, and equipped with GPS receivers for precise orbit and baseline determination. Dramatic improvements in throughput and quality of SAR data processing, both at centers and by individual investigators through research and commercial software packages will increase accessibility of the data and methods to the community, allowing routine exploitation using techniques described above and exploration of new application areas spread across all Earth science disciplines. There are also clear applications of InSAR data from these missions in the commercial sector, in areas such as urban planning, hazard assessment and mitigation, and resource management. These potential commercial applications in turn may provide the drive for more InSAR missions.

Finally, we speculate that this technique will be used beyond the mapping of the earth. It is quite possible to apply this technique to topography mapping in many planetary bodies in the solar system. Although the complexity of the radar systems, the data rates and the required processing power are very challenging, we believe that as we continue our exploration of the solar system, it will be possible to utilize this technique for detailed studies of planetary surfaces. In fact, it is conceivable that the use of the differential interferometric SAR technique will also allow us to investigate the presence of subtle surface changes and probe into the mysteries of the inner working of these bodies.

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APPENDIX

I. SAR Processing Concepts for Interferometry

The precise definition of interferometric baseline and phase and consequently the topographic mapping process depends on how the SAR data are processed. Therefore, before proceeding to the details of topographic mapping using radar interferometry, we give a brief overview of the salient features of SAR data processing.

Processed data from SAR systems are sampled images. Each sample, or pixel, represents some aspect of the physical process of radar backscatter. A resolution element of the imagery is defined by the spectral content of the SAR system. Fine resolution in the range direction is achieved typically by transmitting pulses of either short time duration with high peak power, or of a longer time duration with a wide coded signal bandwidth at lower peak transmit power. Resolution in range is inversely proportional to this bandwidth. In both cases, the received echo for each pulse is sampled at the required radar signal bandwidth. For ultra-narrow pulsing schemes, the pulse width is chosen at the

Fig. 49. The radar emits a sequence of pulses separated in time. The time duration between pulses is called the inter pulse period (IPP) and the associated pulse frequency is called the pulse repetition frequency (PRF=1/IPP). The pulse duration is denoted τ .

desired range resolution, and no further data manipulation is required. For coded pulses, the received echoes are typically processed with a matched filter technique to achieve the desired range resolution. Most spaceborne platforms use chirp-encoding to attain the desired bandwidth and consequent range resolution, where the frequency is linearly changed across the pulse as illustrated in Fig. 49.

Resolution in the azimuth, or along-track, direction, parallel to the direction of motion, is achieved by synthesizing a large antenna from the echoes received from the sequence of pulses illuminating a target. The pulses in the synthetic aperture contain an unfocussed record of the target's amplitude and phase history. To focus the image in azimuth, a digital "lens" that mimics the imaging process is constructed, and is applied by matched filtering. Azimuth resolution is limited by the size of the synthetic aperture, which is governed by the amount of time a target remains in the radar beam. The azimuth beamwidth of an antenna is given by $\theta_{\rm bw} \equiv k\lambda/L$, where λ is the wavelength, L is the antenna length, and k is a constant that depends on the antenna (k=1 is assumed in this paper). The size of the antenna footprint on the ground in the azimuth direction is approximately given by

$$l_{\rm az} = \rho \theta_{\rm bw} = \rho \frac{\lambda}{L} \tag{76}$$

where ρ is the range to a point in the footprint as depicted in Fig. 50.

Fig. 50. The antenna footprint size in the azimuth direction depends on the range and the antenna beamwidth in the azimuth direction.

During the time a target is in the beam, the range and angular direction to the target are changing from pulse to pulse, as shown in Fig. 51. To generate a SAR image, a unique range or angle must be selected from the family of ranges and angles to use as a reference for focusing the image. Once selected, the target's azimuth and range position in the processed image is uniquely established. Specifying an angle for processing is equivalent to choosing a reference Doppler frequency. The bold dashed line from pulse N-2 to the target in Fig. 51 indicates the desired angle or Doppler frequency at which the target will be imaged. This selection implicitly specifies the time of imaging, and therefore the location of the radar antenna. This is an important and often ignored consideration in defining the

interferometric baseline. The baseline, that is the location of the radar antennas forming the interferometry, depends on the choice of processing parameters. Especially for repeat track geometries where the antenna pointing can be different from track to track, careful attention to the baseline model is essential for accurate mapping performance.

Fig. 51. A sensor imaging a fixed point on the ground from a number of pulses in a synthetic aperture. The range at which a target appears in an synthetic aperture image depends on the processing parameters and algorithm used to generate the image. For standard range/Doppler processing the range is fixed by choosing the pulse which has a user defined fixed angle between the velocity vector and the line-of-sight vector to the target. This is equivalent to picking the Doppler frequency.

II. ATMOSPHERIC EFFECTS

For interferometric SAR systems that obtain measurements at two apertures nearly simultaneously, propagation through the atmosphere has two effects which influence interferometric height recovery: 1) delay of the radar signal; and, 2) bending of the propagation path away from a straight line. In practice, for medium resolution IFSAR systems, the first effect dominates.

The atmospheric index of refraction can be written as

$$n(z) = 1 + \delta(z) \tag{77}$$

where z is the height above see level, and $\delta(z)$ represents the variation of the index of refraction as a function of height, and is typically of the order of 10^{-4} . As an example, the CPRL exponential reference atmosphere [121] is given by $\delta(z) = a \exp[-z/H]$ where a and H are given by 3.13×10^{-4} and 6.949 km, respectively.

Rodríguez et al. [17] showed that the relationship between the geometric range ρ and the path distance $\check{\rho}$ is

$$\check{\rho} = \rho \left[1 + \bar{\delta} - \frac{1}{2} \sigma_{\delta}^2 \left[\left(\frac{\rho}{\Delta h} \right)^2 - 1 \right] \right] \tag{78}$$

where $\bar{\delta}$ and σ_{δ}^2 correspond to the height-dependent mean and variance of the variations of the index of refraction, respectively. These two quantities are functions of the height difference between the scatterer and the receiver, Δh , and the height of the scatterer above sea level, h_0 . Using the CPRL model, it is easily seen that the bulk of the effect is dominated by $\bar{\delta}$, i.e., by the mean speed of light in the medium, and produces a fractional error in the range on the order of 10^{-4} if left uncorrected. Corrections based on simple models, such as the CPRL exponential atmosphere, can account for most of the effect and are straightforward to implement.

In a similar way, the interferometric phase can be approximated by

$$\phi \approx -k\Delta\rho(1+\bar{\delta}) + k\rho(\delta(h)-\bar{\delta})\frac{\delta h}{\Delta h}$$
 (79)

where $\Delta \rho$ is the geometric range difference in the path lengths to the IFSAR antennas, and δh is the height separation between the two antennas. At first sight, it might seem as if the last term can be neglected. However, this is not always the case since it is multiplied by the range, which is a large factor.

The results above show that, if one accounts for the mean speed of light of the atmosphere, atmospheric effects will be largely accounted for in single-pass interferometry. This is not the case for repeat-pass interferometry since the atmospheric delays can be different for each pass, and the phase can be dominated by tropospheric variations.

TABLE B-1: SYMBOLS

Symbol	Definition
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λ	Radar wavelength
$ ho_i$	Radar range from ith antenna phase center to a target
$\phi_{ ext{npp}}$	Phase difference between radar channels in standard, or "non-ping-pong" measurement mode
$\phi_{ extsf{pp}}$	Phase difference between radar channels in "ping-pong" measurement mode
p	1 for standard mode, 2 for ping-pong mode
PRF	Radar Pulse Repetition Frequency
IPP	Radar Inter-Pulse Period
RF	Radio Frequency
L	Radar antenna length in the along track direction
$ heta_{ extbf{bw}}$	Radar azimuth beamwidth
T	Target position vector from arbitrary reference
$ec{l_i}$	Look vector, measured from ith platform position reference to target
$\hat{l}_{m{i}}$	Unit look vector
$egin{aligned} heta_{ extbf{bw}} \ ec{T} \ ec{l}_i \ ec{\hat{l}}_i \ ec{P}_i \ ec{B} \ \hat{B} \end{aligned}$	Position vector of ith platform reference, such as antenna phase centers or airframe center of ma
$ec{ec{B}}$	Baseline vector, from phase center of one interferometric aperture to another
Â	Unit vector in direction of baseline
B	Magnitude of the baseline vector
ϕ	general interferometric phase difference between radar channels
α	baseline angle measured counterclockwise from a horizontal reference
$\overset{\backsim}{ heta}$	Look angle measured counterclockwise from nadir
h_0	Surface of constant elevation
\hat{l}_{0}	Unit look vector to a point on constant height reference surface
θ_0	Look angle to a point on constant height reference surface
ρ_0	Radar range to a point on constant height reference surface
ϕ_{flat}	Interferometric phase with phase of constant reference surface removed
h_p	Platform height above spherical Earth of radius r_e
h_t^p	Target height above spherical Earth of radius r_e
$ au_c$	Local terrain slope in cross track direction
i	Local radar incidence angle relative to a spherical Earth
Δho	Range resolution
Δho_{\perp}	Range resolution projected perpendicular to look direction
$B_{\perp, m crit}$	Critical perpendicular baseline, beyond which complete decorrelation
f	Radar frequency
$ec{v}$	Platform velocity
\hat{v}	Unit vector in direction of platform velocity
v	Magnitude of platform velocity

Symbol Definition

TABLE B-2: SYMBOLS

y_0	Cross-track target ground coordinate
$\overset{y_0}{ec{\mathcal{D}}}$	Surface displacement vector
t_i	Time instant i
s	Along-track, or azimuth, coordinate position
c	Across-track coordinate position
h	Vertical coordinate position
\boldsymbol{z}	Topographic height above the reference surface
δho	Surface displacement between imaging times
Δ	Differential operator applied in sensitivity analysis
$E_{\pmb{i}}$	Complex electric field strength at antenna i
γ	Interferometric correlation
γ_N	Thermal noise correlation contribution
γ_G	Geometric correlation contribution
γ_z	Volumetric correlation contribution
$egin{array}{l} \gamma_Z \ \sigma_{m{\phi}}^2 \ N_L \end{array}$	Interferometric phase variance
N_L	Number of independent radar looks
SNR_i	Signal to noise ratio of ith radar channel
W_i	Impulse response of ith radar channel
k	Radar wavenumber
$\frac{\delta k}{2}$	Wavenumber shift
$\delta_{ ho}$	Range error in interferometric image registration
δ_s	Azimuth error in interferometric image registration
$ au_s$	Terrain slope in azimuth direction Interferometric fringe wavenumber projected in range
$\kappa_ ho \ \kappa_z$	Interferometric fringe wavenumber projected in range
f(z)	Effective scatterer probability density function
$\sigma(z)$	Effective normalized backscatter cross-section per unit height
n(z)	Refractive index of atmosphere
$\delta(z)$	Variation in refractive index of atmosphere
\boldsymbol{a}	Reference atmosphere refractivity at the surface of the Earth
H	Reference atmosphere scale height
$\check{ ho}$	Range path length
$reve{\delta}$	Mean of refractivity variations
σ_δ^2	Variance of refractivity variations
Δh	Height difference between target and antenna
δh	Height difference between two antennas
$\mathbf{W}()$	Phase wrapping operator
$rac{\Delta_{i,j}^{x,y}}{ ho_{i,j}^{x,y}}$	Wrapped difference of wrapped phase
$ ho_{i,j}^{x,y}$	Source function consisting of wrapped phase differences
$p_i(t)$	Transmitted pulse from channel i
$h_i(t)$	Baseband transmitted waveform
f_0	Carrier frequency in Hertz
ω_0	Radian carrier frequency
t	Time variable

TABLE B-3: SYMBOLS

t_{di}	Round-trip propagaion time delay of radar signal from antenna i to target to antenna i
t_{d0}	Round-trip propagaion time delay of radar signal from antenna i to target to reference track
$s_i(t)$	Received signal of channel i
$t_{di,\mathrm{REF}}$	Round-trip propagation time delay of radar signal from antenna i to reference elevation h_0
	associated with target, then to antenna i
$t_{d0,\mathrm{REF}}$	Round-trip propagation time delay of radar signal from antenna i to reference elevation h_0
	associated with target, then to reference track
$g_i(t)$	Received signal of channel i resampled to reference track
$c(t,f_0)$	Interferometric correlation function
f_+	Upper sub-band carrier frequency
f_{-}	Lower sub-band carrier frequency
Δf	Sub-band carrier frequency difference
$\phi_{\pm}(t)$	Phase difference of sub-band interferograms
$\phi_0(t)$	Unwrapped phase of interferogram at carrier f_0
$\phi_{ ext{unw}}(t)$	Unwrapped phase on correct absolute ambiguity
n_{α}	number of cycles to add to adjust to absolute phase
$g_j^{\mathcal{S}}(t)$	Received signal of channel j resampled using unwrapped phase to adjust for
	topographic time delay variability
$\phi_i(t)$	Phase measured at output of receiver i
$\omega_{ extbf{BB}}$	Baseband carrier frequency
$ au_{r}^{i})$	Total time delay through receiver i
$ au_{m{k}}^i$	Delay in the k th filter/amplifier in receiver i
$\omega_{m{k}}$	Frequency of the k th mixer in receiver chain
$ au_i$	Delay associated with antenna i amplifier electronics
ϕ_k^i	Phase reference of the k th mixer in receiver i
$\phi_{ m T}$	Total transmitter phase delay
$ au_{ extbf{T}}$	Total transmitter time delay

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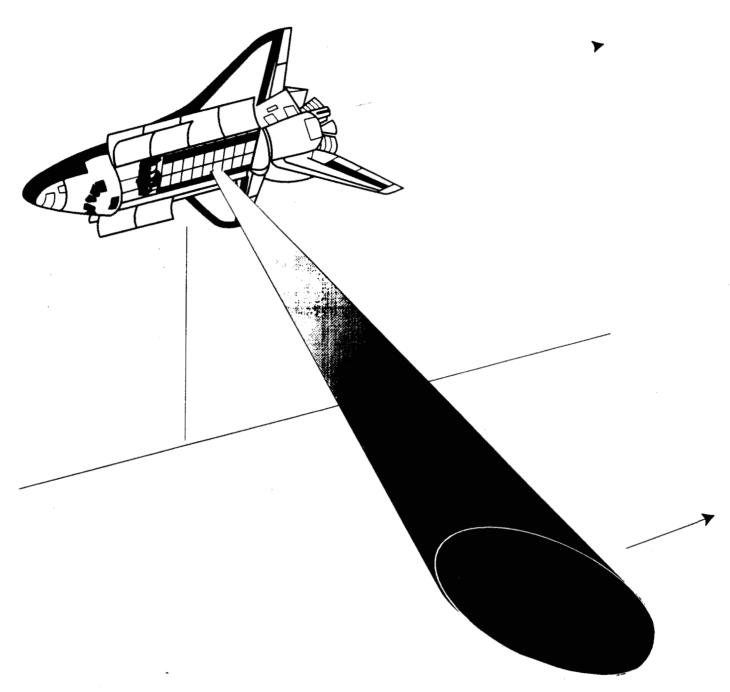


Figure 1: Imaging scenario for typical SAR system. The platform carrying the SAR instrument follows a rectilinear track. The radar antenna looks to the side, imaging the terrain below.

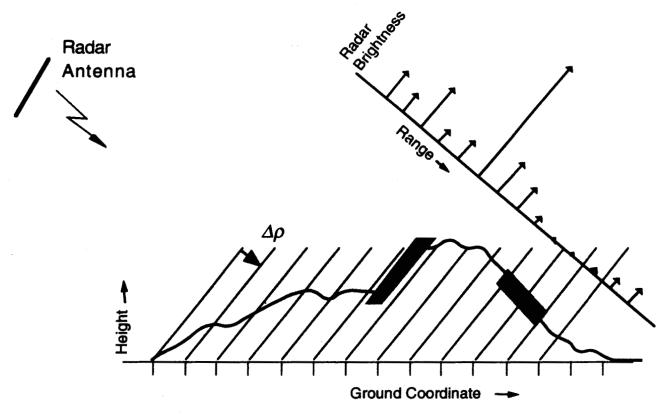


Figure 2: The three-dimensional world is collapsed to two dimensions in conventional SAR imaging.

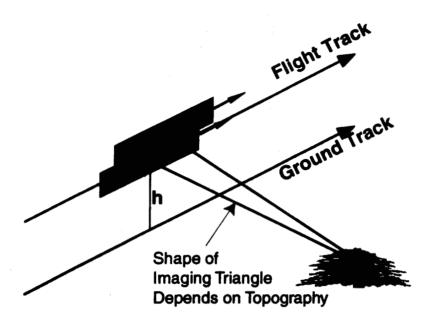


Figure 3: Interferometric SAR for topographic mapping uses two apertures separated by a "baseline" to image the surface. The phase difference between the apertures for each image point, along with the range and knowledge of the baseline, can be used to infer the precise shape of the imaging triangle to derive the topographic height of the image point.

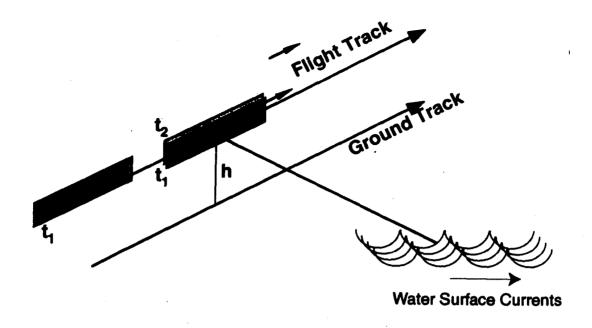


Figure 4: An along track interferometer maintains a baseline separated along the flight track such that surface points are imaged by each aperture within one second. Motion of the surface over the elapsed time is recorded in the phase difference of the pixels.

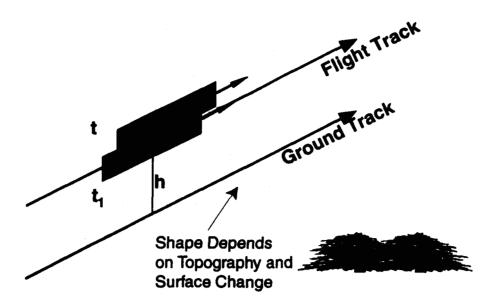


Figure 5: A repeat track interferometer is similar to an along track interferometer. An aperture repeats its track and precisely measures motion of the surface between observations in the image phase difference. If the track does not repeat at exactly the same location, some topographic phase will also be present, which must be removed by the methods of differential interferometry to isolate the motion.

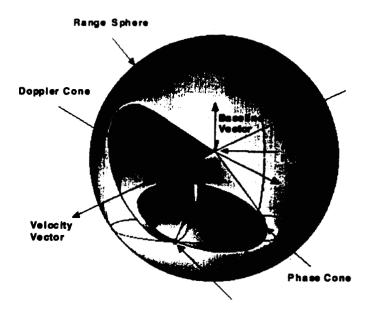


Figure 6: Target location in an IFSAR image is precisely determined by noting that the target location is the intersection of the range sphere, doppler cone and phase cone.

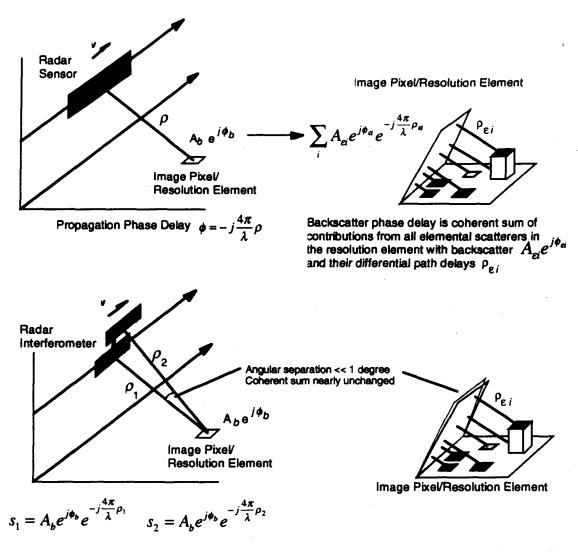


Figure 7: The interferometric phase difference is mostly due to the propagation delay difference. The (nearly) identical coherent phase from the different scatterers inside a resolution cell (mostly) cancels out during interferogram formation.

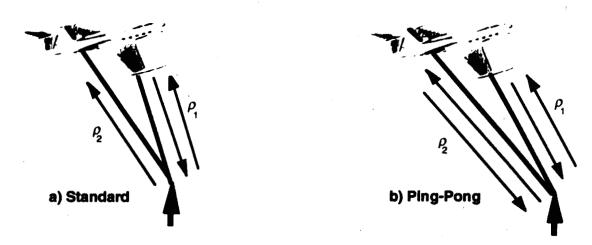


Figure 8: Illustration of standard vs. ping-pong mode of data collection. In standard mode, the radar transmits a signal out of one of the interferometric antennas only, and receives the echoes through both antennas A_1 and A_2 simultaneously. In "ping-pong" mode, the radar transmits alternatively out of the top and bottom antennas and receives the radar echo only through the same antenna.

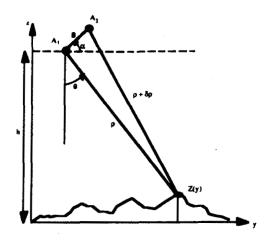
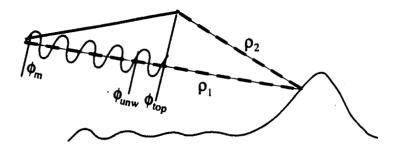


Figure 9: SAR interferometry imaging geometry.



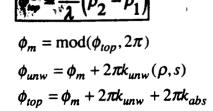


Figure 10: Phase measurement in the interferogram is intially known modulo 2π . After unwrapping phase relative phase measurements between all pixels in the interferogram are determined up to a constant multiple of 2π Absolute phase determination is the process to determine the overall multiple of 2π that must be added to the phase measurements to that the have the correct proportionality to the range difference.

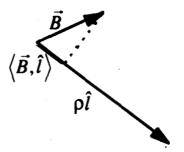


Figure 11: When the plane wave approximation is valid the range difference is approximately the projection of the baseline vector onto a unit vector in the line of sight direction.

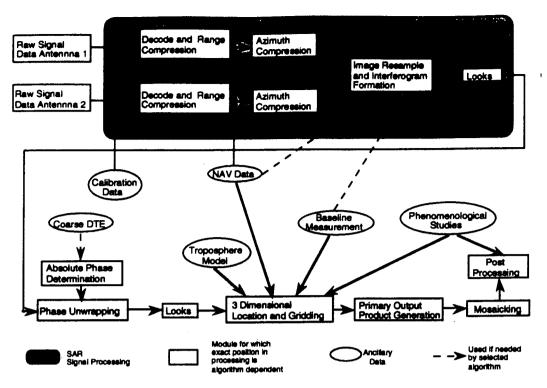


Figure 13: Block diagram showing the major steps in interferometric processing to generate topographic maps. Data for each interferometric channel are processed to full resolution images using the platform motion information to compensate the data for perturbations from a straight line path. One of the complex images is resampled to overlay the other, and an interferogram is formed by cross-multiplying images, one of which is conjugated. The resulting interferogram is averaged to reduce noise. Then, the principal value of the phase for each complex sample is computed. To generate a continuous height map, the two-dimensional phase field must be unwrapped. After the unwrapping process, an absolute phase constant is determined. Subsequently, the three dimensional target location is performed with corrections applied to account for tropospheric effects. A relief map is generated in a natural coordinate system aligned with the flight path. Gridded products may include the target heights, the SAR image, a correlation map, and a height error map.

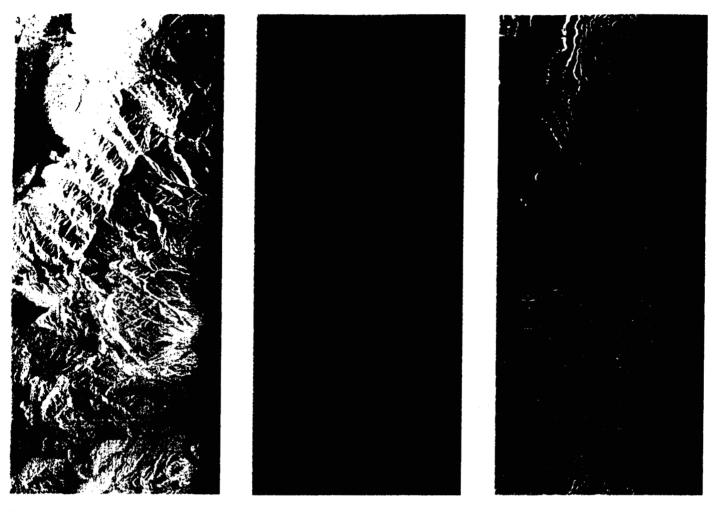


Figure 12: (a) Radar brightness image of Mojave desert near Fort Irwin, California. (b) Interferogram of the area showing intrinsic fringe variability (c) Flattened interferogram assuming a reference surface at zero elevation above a spherical earth.

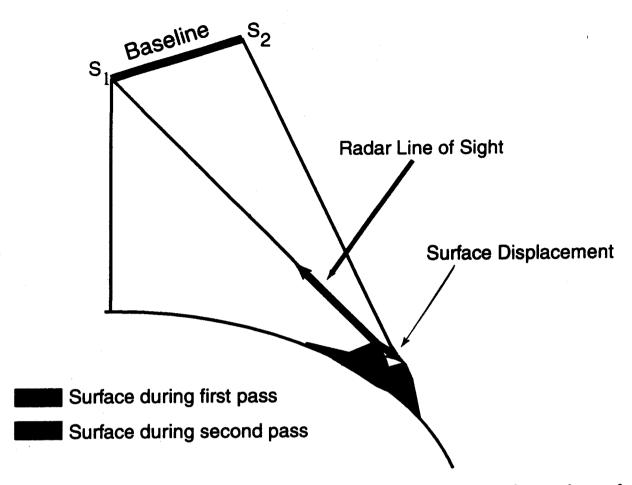


Figure 14: Geometry of displacement interferometry. Surface element has moved in a coherent fashion between observation A_1 made at time t_1 and observation A_2 made at time t_2 . The displacement can be of any sort - continuous or instantaneous, steady or variable - but the detailed scatterer arrangement must be preserved in the interval for coherent observation.

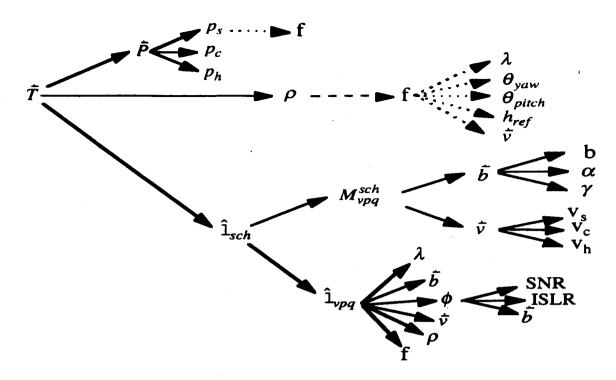


Figure 15: Sensitivity tree showing the sensitivity of target location to various parameters used in interferometric height reconstruction. A dotted line denotes a dependency that is processing or algorithm dependent.

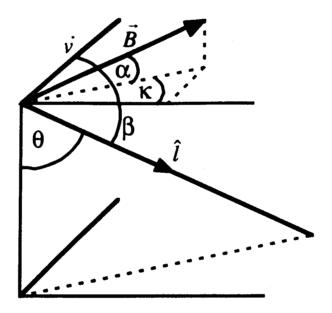


Figure 16: Baseline and look angle geometry as used in sensitivity formulas.

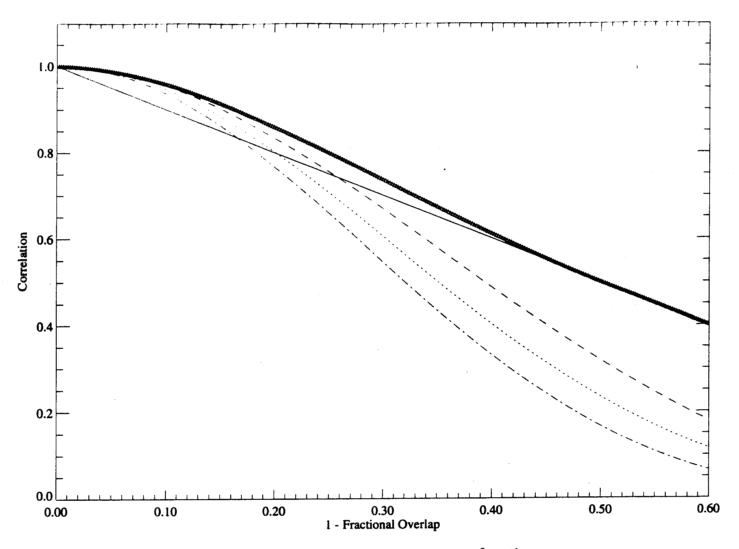


Figure 20: Baseline decorrelation for various point target response functions.

INTERFEROMETER DECORRELATION ANGLE PROPORTIONAL TO WAVELENGTH'L

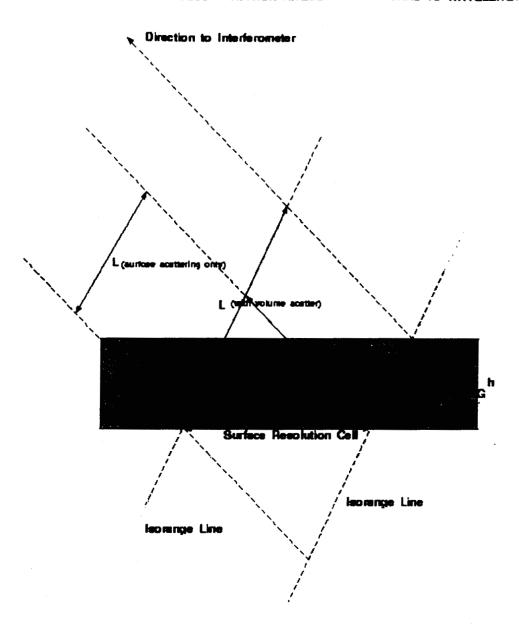


Figure 21: Form of the volumetric decorrelation.

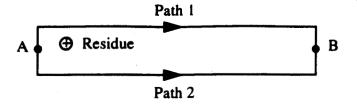
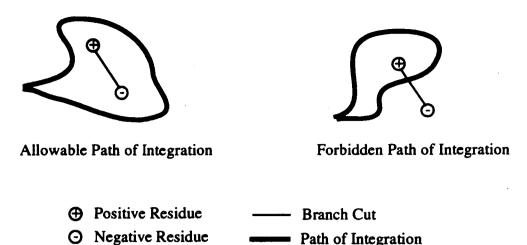


Figure 17: An example of two possible paths of integration from A to B with a nearby residue.



- Path of Integration

Figure 18: An example of a branch cut and allowable and forbidden paths of integration.

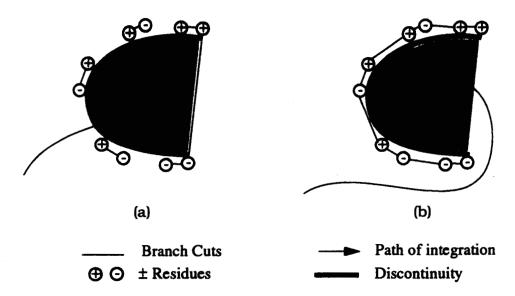


Figure 19: Cut dependencies of unwrapped phase. a) shortest path cuts b) better choice of cuts.

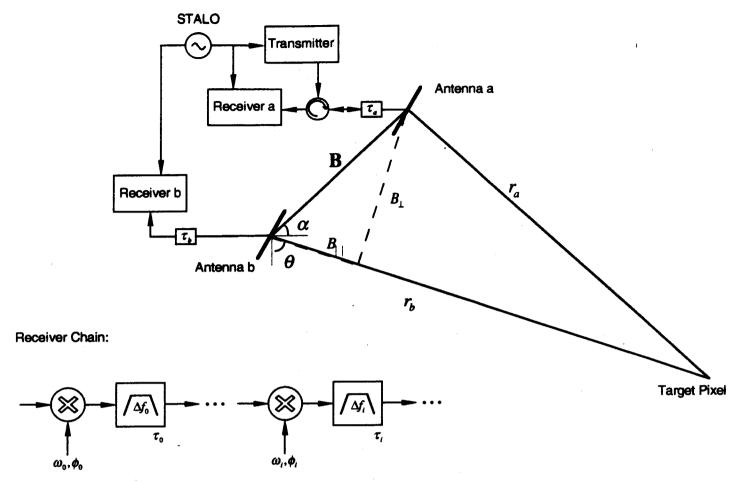


Figure 22: Definitions of interferometric parameters relating to a possible radar interferometer configuration. In this example, the transmitter path is common to both roundtrip signal paths. Therefore the transmitter phase and time delays cancel in the channel difference.

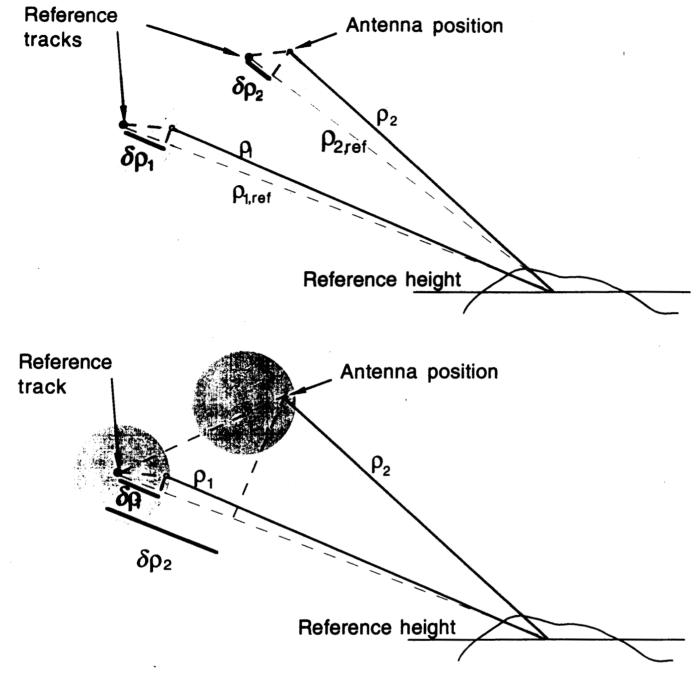


Figure 23: Motion compensation geometry illustrated for interferometry. a) Two flight paths and their idealized companions. b) Motion compensation formulation for a dual reference track approach.

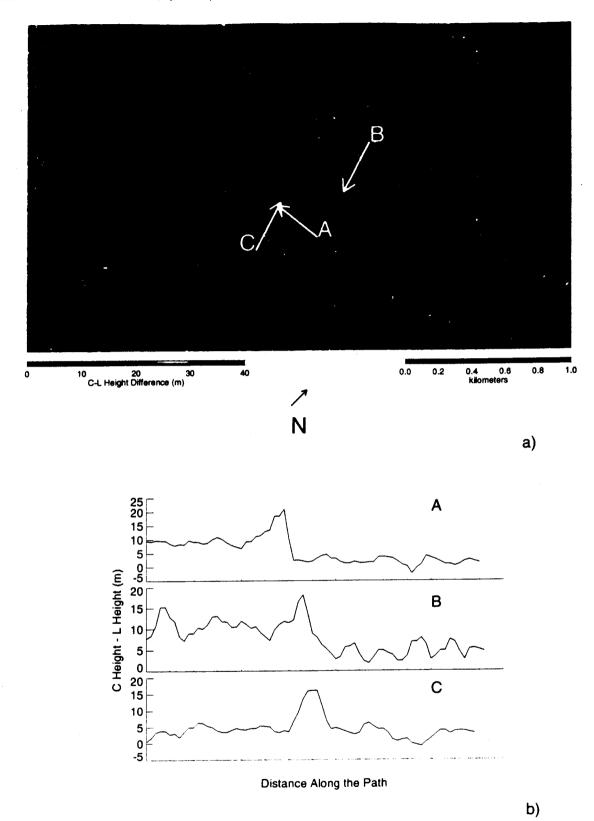


Figure 24: a) Image of height difference between C- and L-band in Iron Mountain, California. b) Profiles as indicated going from bare fields to forested regions.

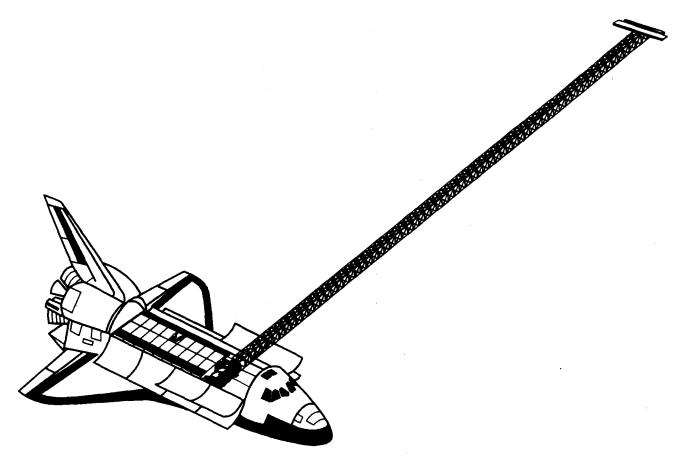


Figure 25: The Shuttle Radar Topography Mission flight system configuration. The SIR-C/X-SAR L, C, and X-band antennas reside in the shuttle's cargo bay. The C and X band radar systems are augmented by receive-only antennas deployed at the end of a 60 m long boom. Interferometric baseline length and attitude measurement devices are mounted on a plate attached to the main L-band antenna structure. During mapping operations, the shuttle is oriented to that the boom is 45 degrees from the horizontal.

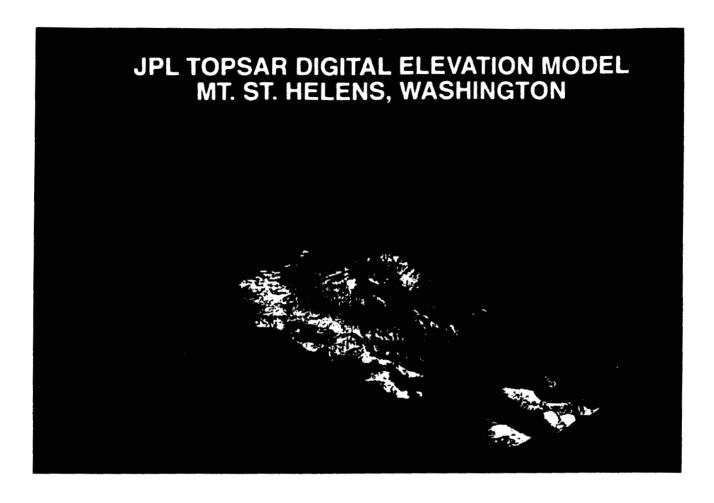
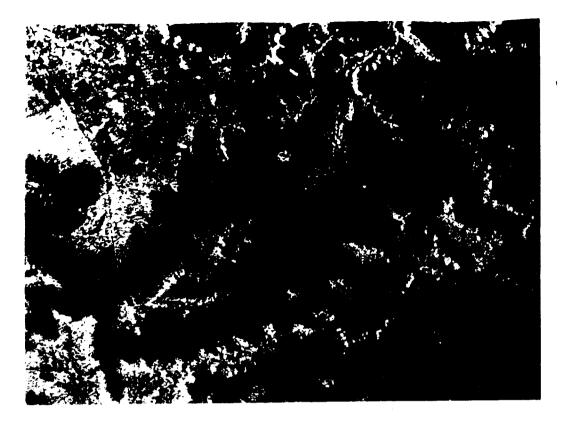


Figure 26: DEM of Mount Saint Helens generated in 1992 with the TOPSAR C-band interferometer. Area covered is roughly 6 km across track by 20 km along track.



Figure 27: DEM of Askja, Northern volcanic zone, Iceland derived from the C-band EMISAR topographic mapping system. The color variation in the image is derived from L-band EMISAR polarimetry.



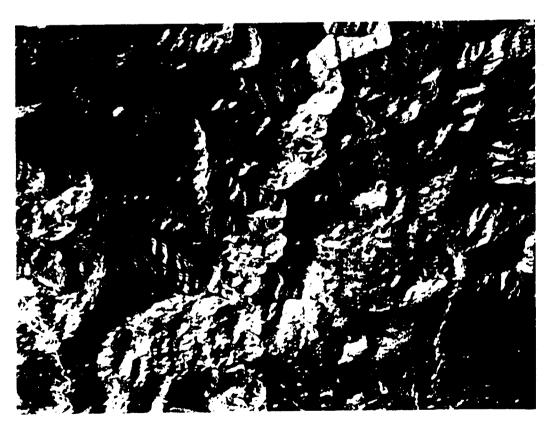


Figure 28: Height suboversand slope shelow emaps of Mount Sonoma, CA. This information has been used to assess the risk of earthquake damage induced by landshelo.

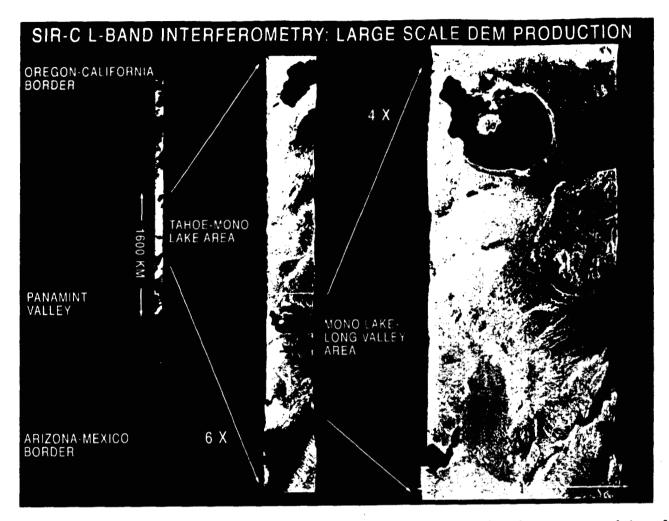


Figure 29: Strip of Topography generated from the SIR-C L-band radar data by repeat track interferometry. The DEM extends from the Oregon/California border through California to Mexico, roughly 1600 km.

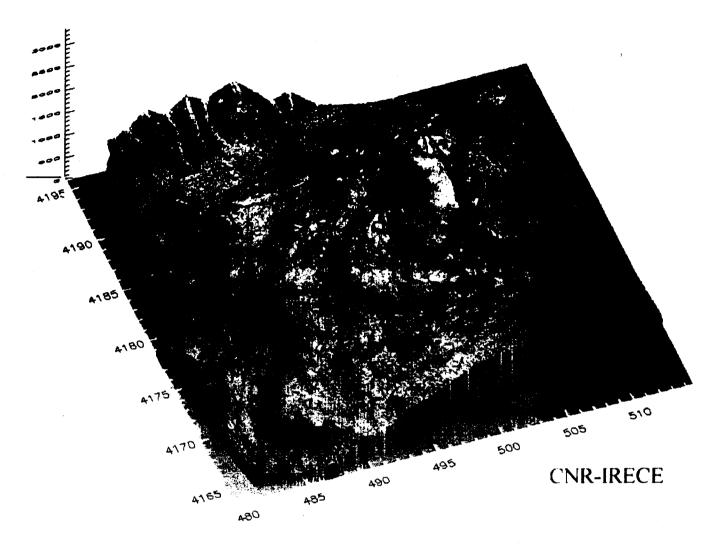


Figure 30: DEM of Mount Etna, Italy generated by ERS repeat track interferometry. Actually 10 images were combined to make this DEM.



Figure 31: DEM of Mount Unzen, Japan generated by JERS repeat track interferometry.



Figure 32: Long Valley mosaic of TOPSAR C-band interferometric data.

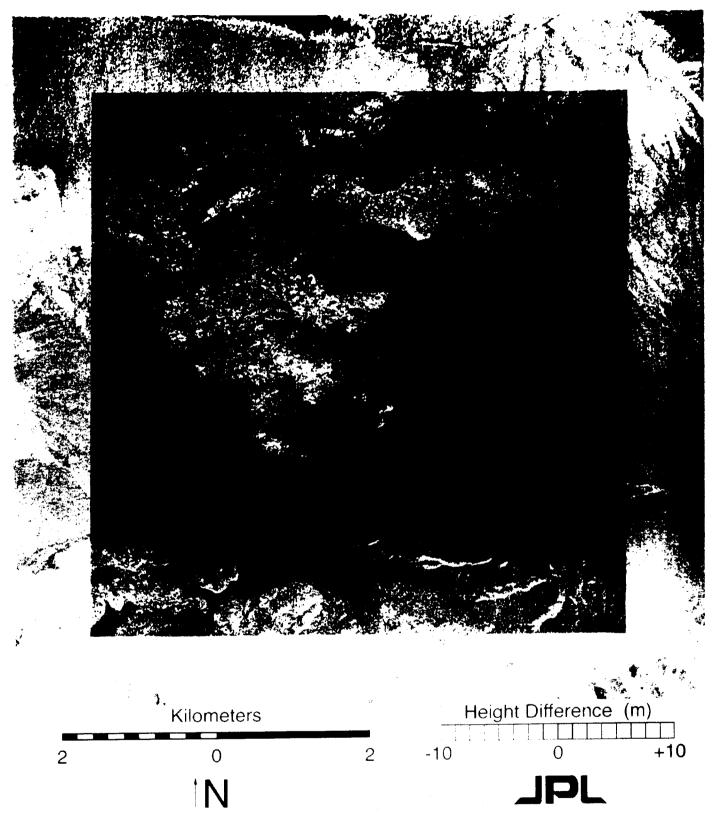


Figure 33: Difference image between TOPSAR C band derived DFM and a TEC photogrammetrically-generated reference DEM

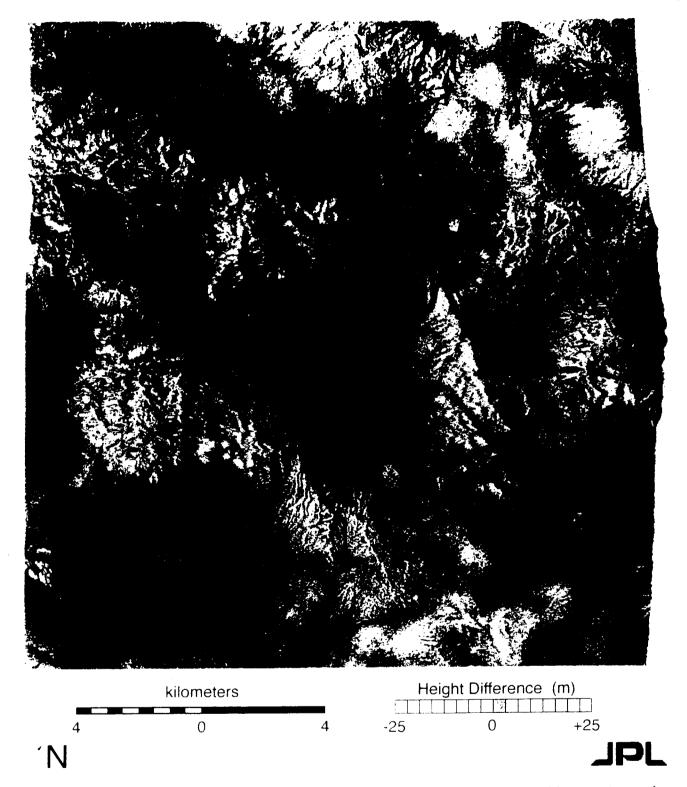


Figure 34 Difference image between SIR-C C band derived DEM and a TOPSAR mosaic used as a reference DEM.

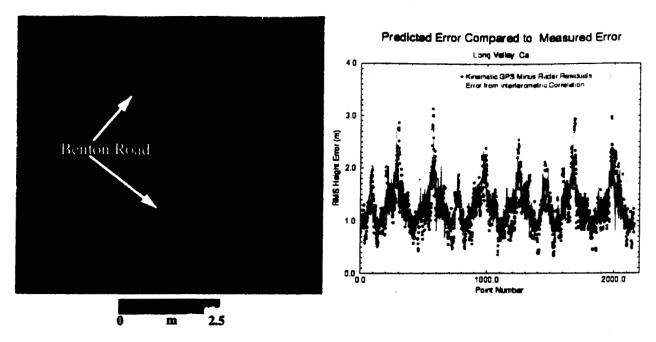


Figure 35: Illustration of the use of kinematic GPS surveys in determining the absolute and relative error of a radar-derived DEM. Curve shows the standard deviation of the radar height relative to the GPS, and its predicted value. Statistical height error estimates derived from the correlation track the measured local statistical height errors extremely well [70].

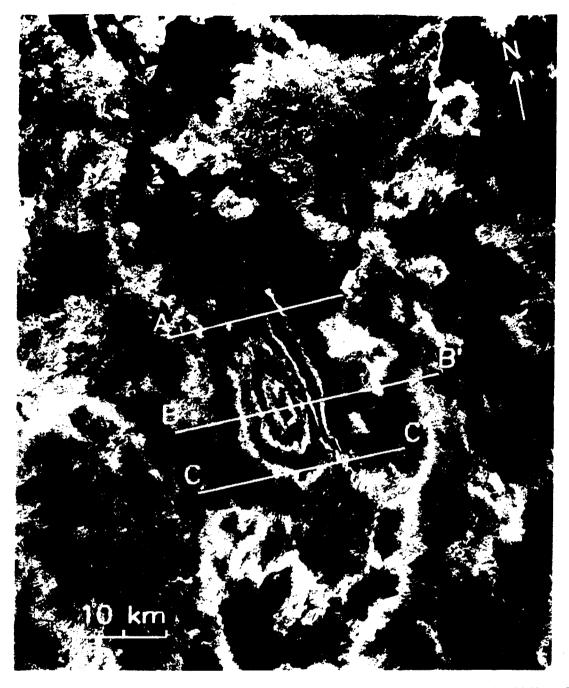


Figure 36: Subsidence caused to an M=6.3 earthquake along a normal fault in Eureka Valley, California imaged interferometrically by ERS-1. The interferometric signature combined with the seismic record allowed unique interpretation of variable slip along the fault.

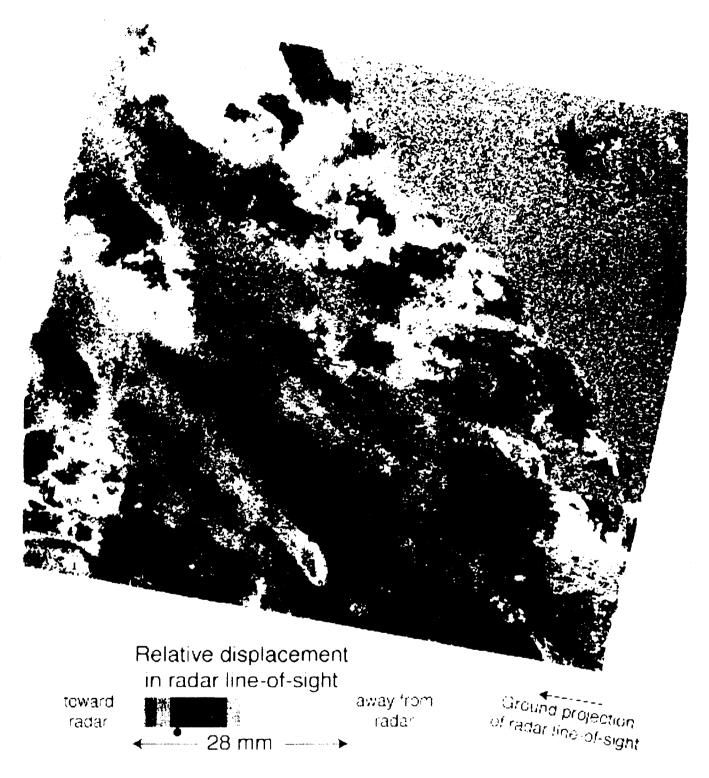


Figure 37: Aseismic slip along the San Andreas Fault near Parkfield, California, imaged interferometrically by ERS-1.

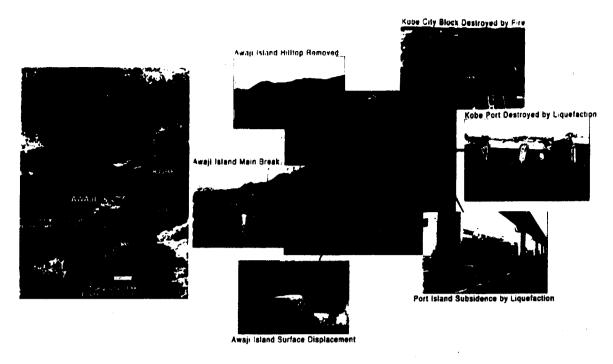


Figure 38: Decorrelation in the destroyed areas of Kobe city due to the 1995 M=6.8 earthquake. Areas where structures were firmly connected to bedrock remained correlated, while structures on sandy areas of liquefaction were destroyed and decorrelated in the imagery.

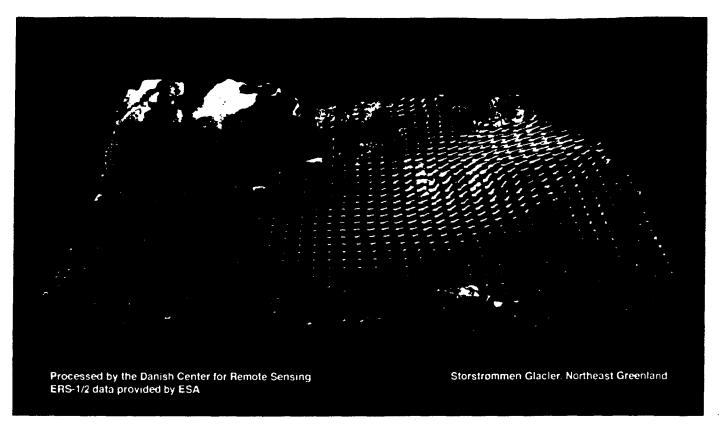


Figure 39: Ice velocity map draped on topography of Storstrømmen Glacier in Greenland. Both velocity and topography were generated by ERS interferometry. Ice velocity vectors show that the outlet of the glacier is blocked from flow [104]. In addition to aiding visualization of the ice flow, topographic maps such as this are an important measurement constraint on the mass balance, as changes in topographic height relate to the flow rate of ice from the glacier to the sea.

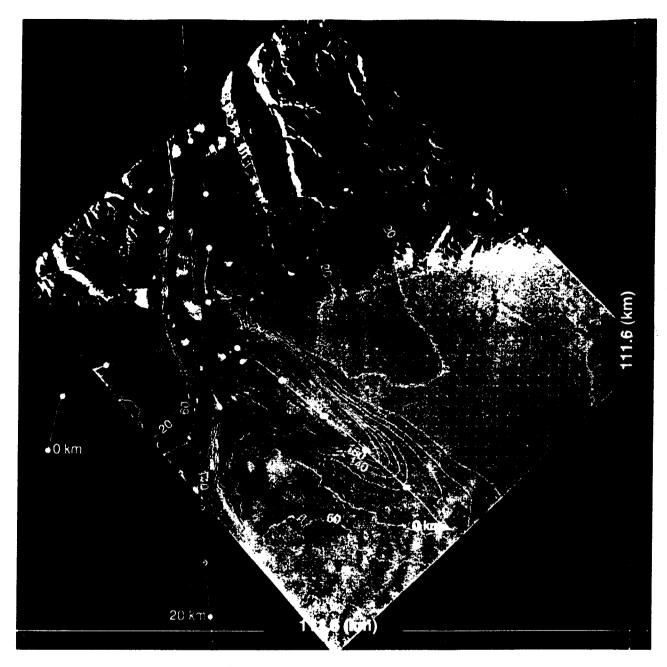


Figure 40: Horizontal velocity field plotted over the SAR amplitude image of the Ryder Glacier. Contour interval is 20 m/yr (cyan) for velocity less than 200 m/yr and is 100 m/yr (blue) for values greater than 200 m/yr. Red arrows indicate flow direction and have length proportional to speed.

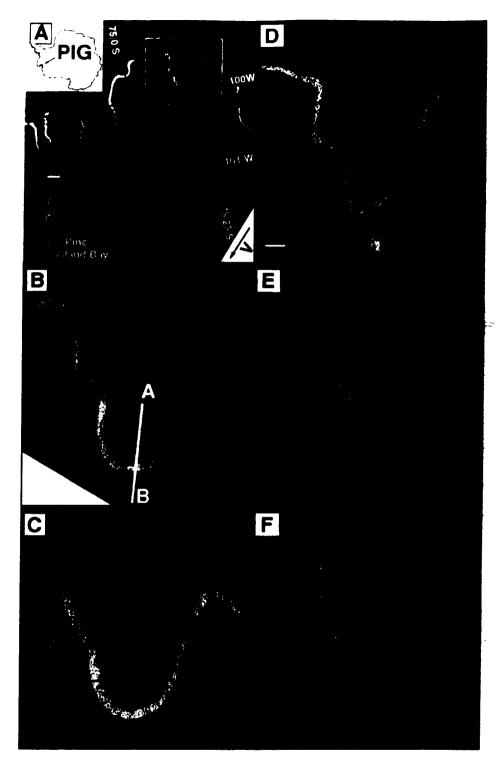


Figure 41: Grounding line time series (Courtesy: E. Rignot; Copyright Science), illustrating the retreat of Pine Island Glacier.

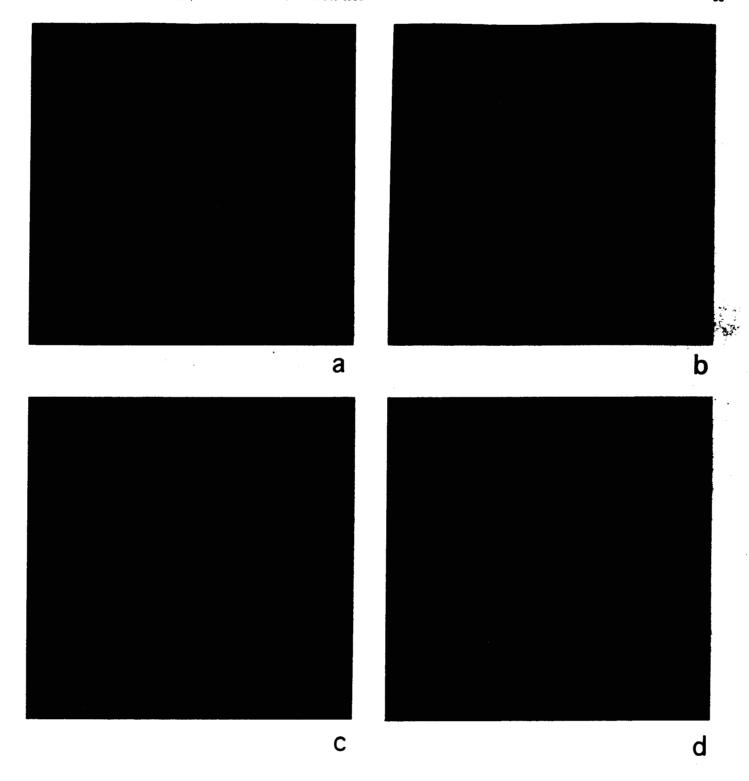


Figure 42: Example of ocean currents measured by along-track SAR interferometry at Mission Bay, San Diego, CA. Flight direction of the radar is from left to right in each image, so panels a-d show different look aspects of the wave patterns propagating to shore.

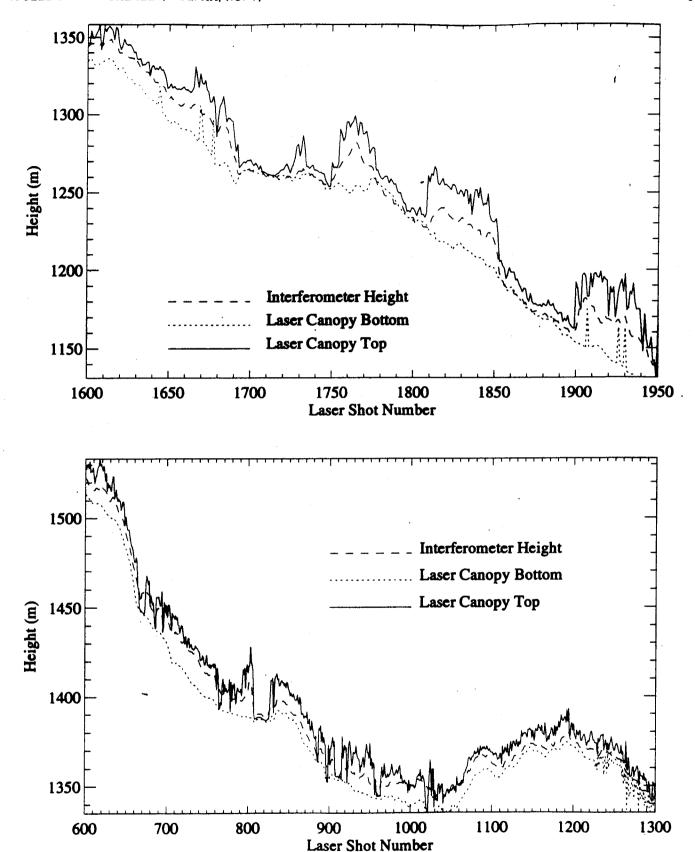


Figure 43: Profiles of canopy extent as measured by the Goddard Space Flight Center Airborne Laser Altimeter.

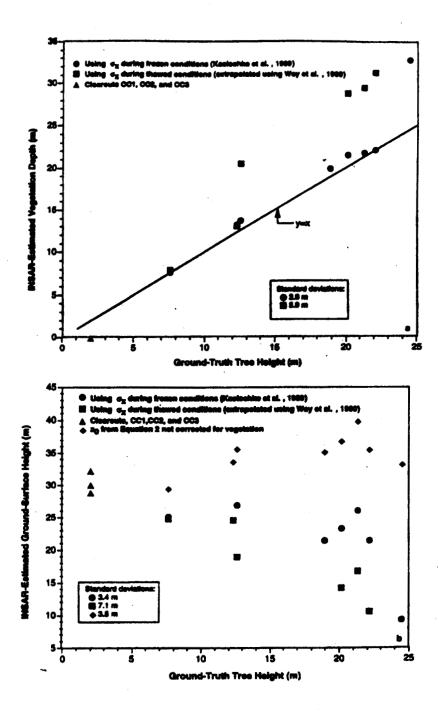


Figure 44: Inversion of tree height and elevation. Courtesy R. Treuhaft. Copyright Radio Science.

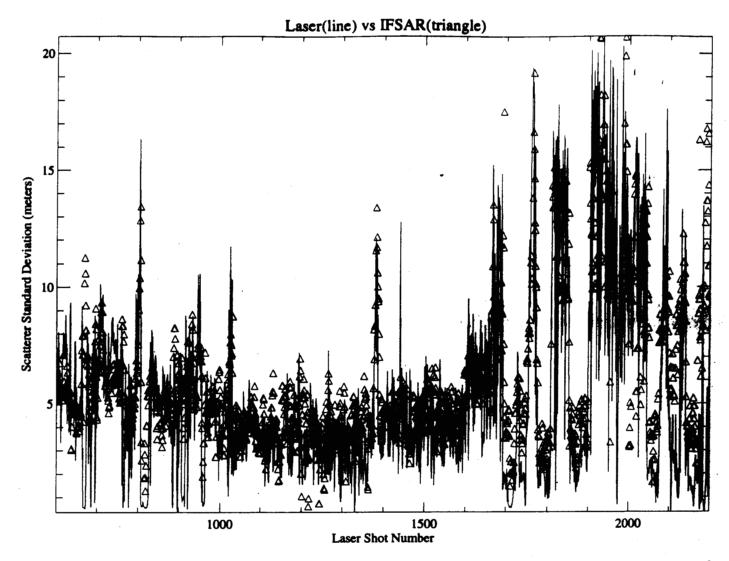


Figure 45: Estimated scatterer standard deviation scaled empirically, compared to tree height derived by laser altimeter. Agreement is very good.

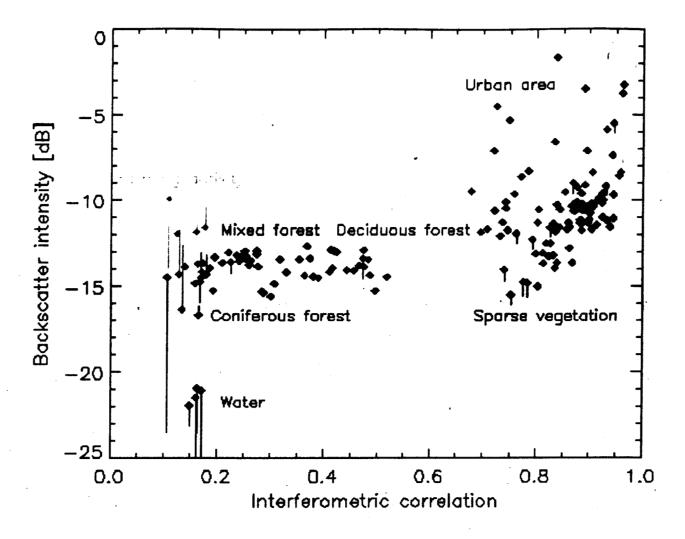


Figure 46: Classification space showing image brightness vs. interferometric correlation. Terrain types cluster as indicated.

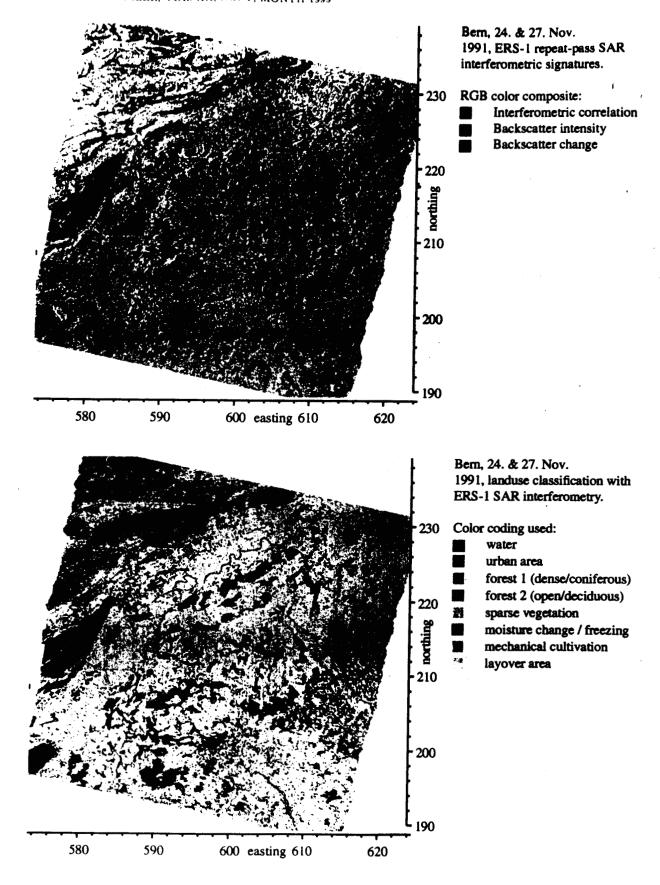


Figure 47: Classification of Bern, Switzerland using L- and C- band SIR-C interferometric time series data to distinguish features.

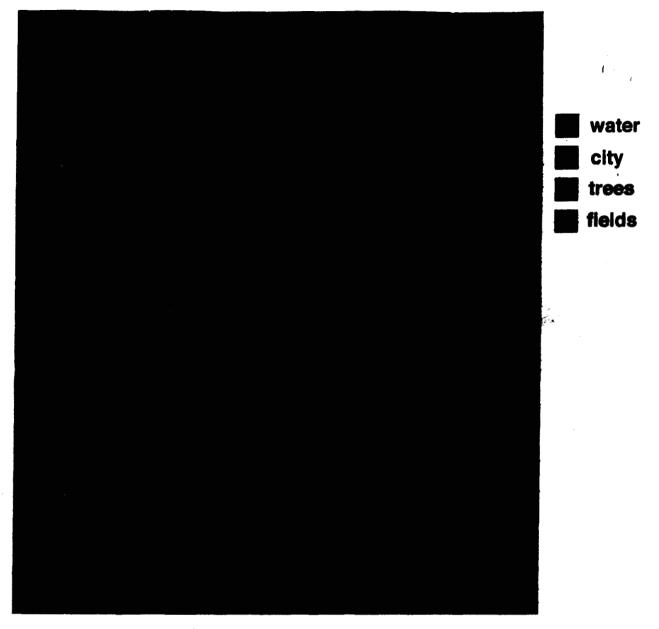


Figure 48: Classification of San Francisco using JPL TOPSAR image brightness, inteferometric correlation, and topographic height and slope.

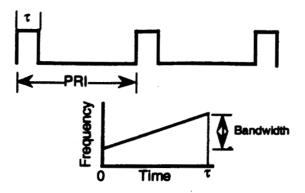


Figure 49: The radar emits a sequence of pulses separated in time. The time duration between pulses is called the inter pulse period (IPP) and the associated pulse frequency is called the pulse repetition frequency (PRF=1/IPP). The pulse duration is denoted τ .

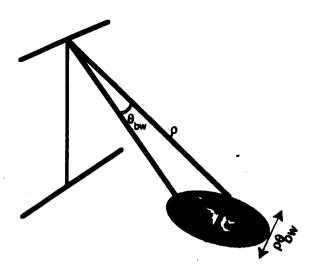


Figure 50: The antenna footprint size in the azimuth direction depends on the range and the antenna beamwidth in the azimuth direction.

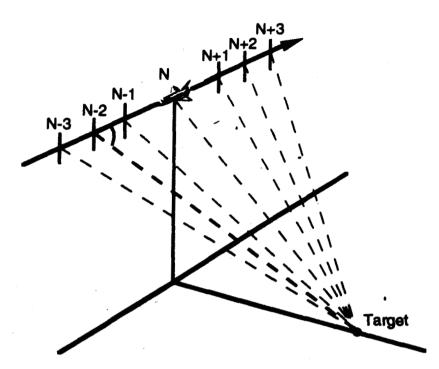


Figure 51: A sensor imaging a fixed point on the ground from a number of pulses in a synthetic aperture. The range at which a target appears in an synthetic aperture image depends on the processing parameters and algorithm used to generate the image. For standard range/Doppler processing the range is fixed by choosing the pulse which has a user defined fixed angle between the velocity vector and the line-of-sight vector to the target. This is equivalent to picking the Doppler frequency.